

Robust LPV-based infinite horizon LQR design*

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Abstract—In this paper, the problem of robust infinite horizon linear quadratic regulator (LQR) design is addressed for uncertain affine linear parameter-varying (LPV) systems. The proposed method extends the standard infinite horizon LQR design to LPV-based static output-feedback (SOF), dynamic output-feedback (DOF) and to a well known proportional, integral and derivative (PID) controller design for uncertain affine LPV systems. The optimal (suboptimal) controller design is formulated as an optimization problem subject to some linear/bilinear matrix inequality (LMI/BMI) constraints. As the main result, the suggested performance and stability conditions, without any restriction on the controller and system structure, are convex functions of the scheduling and uncertainty parameters. Hence, there is no need for applying multi-convexity or other relaxation techniques and consequently the proposed solution delivers a less conservative design method. The viability of the novel design technique is demonstrated and evaluated through numerical examples.

I. INTRODUCTION

One of the most fundamental problems in control theory is the linear quadratic regulator (LQR) design problem [1]. The so-called infinite horizon linear quadratic problem of finding a control function $u^* \in \mathbb{R}^m$ for $x_0 \in \mathbb{R}^n$ that minimizes the cost functional:

$$J^* = \int_0^\infty \left(x(t)^T Q x(t) + u^T(t) R u(t) + 2x^T(t) N u(t) \right) dt, \quad (1)$$

with $R > 0$, $Q - N R^{-1} N^T \geq 0$ subject to $\dot{x}(t) = A x(t) + B u(t)$, $x(0) = x_0$ has been studied by many authors [1], [2], [3], [4]. In many cases, it is not possible or economically feasible to measure all the state variables. This resulted in several generalized versions of the above problem to output-feedback with both necessary and sufficient conditions for the existence of a solution [5], [6], [7]. The well known proportional, integral and derivative (PID) equivalent of a LQR can be also found in [8].

Subsequently, in many papers have been studied the robust static output-feedback version of the LQR design [9], [10], [11], as well as the LQR-based PID controller design [12], [13], [14]. The introduction of the linear parameter-varying (LPV) systems [15] has opened new possibilities in LQR design. Several gain-scheduled/LPV-based LQR design

techniques appeared in both static output feedback (SOF) and dynamic output-feedback (DOF), not to mention the PID controller design [16], [17], [18], [19], [20], [21], [22].

Nonetheless, all of the mentioned approaches have some limitations. The majority of these approaches uses the multi-convexity lemma [23] to obtain a finite number of linear/bilinear matrix inequalities (LMI/BMI). This is introducing some conservativeness in the controller design, since it forces the Bellman-Lyapunov function to be multi-convex. Thus, different relaxation techniques have been deployed to reduce the conservativeness caused by the multi-convexity requirement [18], [19], [24]. Nevertheless, these relaxations can have significant influence on the performance and can drift the guaranteed cost far away from its optima. Multi-convexity has been differently solved, usually by restricting the closed-loop LPV structure, system or controller to avoid cross term effects of the scheduling parameters [21], [22].

From this short literature survey follows that there is no general LPV-based LQR controller design approach, which gives convex dependency on scheduling parameters without major restrictions on the closed-loop system or controller matrices. Therefore, in this paper we present a generalized LPV-based LQR design for uncertain affine LPV systems which allows beside SOF and DOF the well known PID controller structure as well. Furthermore, the suggested performance and stability conditions, without any restriction on the controller and system structure, are convex functions of the scheduling and uncertain parameters. Hence, there is no need for applying the multi-convexity or other relaxation techniques.

The rest of the paper is organized into four sections. The introduction is followed by preliminaries and problem formulation in Section 2. The main result is presented in Section 3, where the generalized LPV-based LQR design is presented. Numerical examples are given in Section 4. Concluding remarks close the paper in Section 5.

The mathematical notation of the paper is as follows. Given a symmetric matrix $P = P^T \in \mathbb{R}^{n \times n}$, the inequality $P > 0$ ($P \geq 0$) denotes the positive definiteness (semi definiteness) of the matrix. Matrices, if not explicitly stated, are assumed to have compatible dimensions. I denotes the identity matrix of corresponding dimensions. The symbol $*$ denotes a block that is readily inferred by symmetry. Notation for interval of numbers between a and b including endpoints a and b is $\langle a, b \rangle = \{x \in \mathbb{R} | a \leq x \leq b\}$.

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II. PRELIMINARIES AND PROBLEM FORMULATION

Consider the following uncertain LPV system as follows:

$$\begin{aligned} E(\theta(t))\dot{x}(t) &= A(\theta(t))x(t) + B(\theta(t))u(t), \\ y(t) &= C(\theta(t))x(t) + D(\theta(t))u(t). \end{aligned} \quad (2)$$

where $x(t) \in \mathbb{R}^n$, $y(t) \in \mathbb{R}^l$, and $u(t) \in \mathbb{R}^m$ are the state, measurable output, and the control input, respectively. The matrix functions $E(\theta(t)) \in \mathbb{R}^{n \times n}$, $A(\theta(t)) \in \mathbb{R}^{n \times n}$, $B(\theta(t)) \in \mathbb{R}^{n \times m}$, $C(\theta(t)) \in \mathbb{R}^{l \times n}$ and $D(\theta(t)) \in \mathbb{R}^{l \times m}$ are assumed to depend on the scheduling variable $\theta(t) \in \langle \underline{\theta}, \bar{\theta} \rangle \in \Omega$ as (3) with $S(\theta(t)) = \{E(\theta(t)), A(\theta(t)), B(\theta(t)), C(\theta(t)), D(\theta(t))\}$.

$$\begin{aligned} S(\theta(k)) &= S_0 + \sum_{i=1}^{N_\alpha} S_i \alpha_i(k) + \sum_{j=1}^{N_\beta} S_{(N_\alpha+j)} \beta_j(k) \\ &+ \sum_{g=1}^{N_\delta} S_{(N_\alpha+N_\beta+g)} \delta_g(k) = S_0 + \sum_{i=1}^p S_i \theta_i(k). \end{aligned} \quad (3)$$

The scheduling variable used in this paper is extended and distributed to:

$$\theta(k) = [\alpha_1, \dots, \alpha_{N_\alpha}, \beta_1, \dots, \beta_{N_\beta}, \delta_1, \dots, \delta_{N_\delta}], \quad (4)$$

where

- a) it is assumed that the scheduling parameters $\alpha_i(k)$ $i = 1, 2, \dots, N_\alpha$ are constant or time-varying, furthermore can be measured or estimated and therefore used in the controller,
- b) the scheduling parameters $\beta_j(k)$, $j = 1, 2, \dots, N_\beta$ are constant or time-varying but unknown (uncertain) parameters,
- c) $\delta_g(k)$, $g = 1, 2, \dots, N_\delta$ represents the uncertain part of the measurable/estimable scheduled parameters (see *Appendix A*),

It is assumed that the maximal rate of change of scheduled parameters $\max \dot{\theta}_i(t) \leq \rho_{\theta_i}$ are known and predefined. Furthermore, the matrix $E(\theta(t))$ is non-singular.

We are looking for a static output-feedback defined as

$$u(t) = F(\theta(t))y(t), \quad (5)$$

where

$$F(\theta(t)) = F_0 + \sum_{i=1}^p F_i \theta_i(t) \in \mathbb{R}^{m \times l}. \quad (6)$$

Notice, that the controller gain matrices F_i related to the uncertain parameters $\beta(t)$ and $\delta(t)$ are equal to zero.

The control law (5) can be augmented with integral part [14] for PI controller design. In this case the augmented system is defined as:

$$\begin{aligned} \bar{E}(\theta(t))\dot{\bar{x}}(t) &= \bar{A}(\theta(t))\bar{x}(t) + \bar{B}(\theta(t))\bar{u}(t), \\ \bar{y}(t) &= \bar{C}(\theta(t))\bar{x}(t) + \bar{D}(\theta(t))\bar{u}(t), \end{aligned} \quad (7)$$

where $\bar{x}(t)^T = [x(t), z(t)] \in \mathbb{R}^{n+l}$ is the augmented state vector, $\bar{y}(t)^T = [y(t), z(t)] \in \mathbb{R}^{l+l}$ is the augmented output

vector, and $\bar{u}(t) = u(t) \in \mathbb{R}^m$ is the augmented input vector. Furthermore, the augmented system and control matrices are:

$$\begin{aligned} \bar{E}(\theta(t)) &= \begin{bmatrix} E(\theta(t)) & 0 \\ 0 & I \end{bmatrix}, \bar{A}(\theta(t)) = \begin{bmatrix} A(\theta(t)) & 0 \\ C(\theta(t)) & 0 \end{bmatrix}, \\ \bar{B}(\theta(t)) &= \begin{bmatrix} B(\theta(t)) \\ 0 \end{bmatrix}, \bar{C}(\theta(t)) = \begin{bmatrix} C(\theta(t)) & 0 \\ 0 & I \end{bmatrix}, \\ \bar{D}(\theta(t)) &= \begin{bmatrix} D(\theta(t)) \\ 0 \end{bmatrix}, \bar{F}(\theta(t)) = [F(\theta(t)), F_i(\theta(t))]. \end{aligned} \quad (8)$$

We can design a full/reduced order dynamic output-feedback controller with order n_k , defined as:

$$\begin{aligned} \dot{x}_c &= A_c(\theta(t))x_c(t) + B_c(\theta(t))y(t), \\ u(t) &= C_c(\theta(t))x_c(t) + D_c(\theta(t))y(t), \end{aligned} \quad (9)$$

$$\begin{aligned} x_c(t) &\in \mathbb{R}^{n_k}, A_c(\theta(t)) \in \mathbb{R}^{n_k \times n_k}, B_c(\theta(t)) \in \mathbb{R}^{n_k \times l}, \\ C_c(\theta(t)) &\in \mathbb{R}^{m \times n_k}, D_c(\theta(t)) \in \mathbb{R}^{m \times l}, \end{aligned}$$

since the system (2) can be reformulated to static output-feedback design, too [25]. In this case, the controller gain matrices affinely depend on the scheduling variable as (3) with $S(\theta(t)) = \{A_c(\theta(t)), B_c(\theta(t)), C_c(\theta(t)), D_c(\theta(t))\}$ and the augmented system matrices are as follows:

$$\begin{aligned} \bar{E}(\theta(t)) &= \begin{bmatrix} E(\theta(t)) & 0 \\ 0 & I \end{bmatrix}, \bar{A}(\theta(t)) = \begin{bmatrix} A(\theta(t)) & 0 \\ 0 & 0 \end{bmatrix}, \\ \bar{B}(\theta(t)) &= \begin{bmatrix} B(\theta(t)) & 0 \\ 0 & I \end{bmatrix}, \bar{C}(\theta(t)) = \begin{bmatrix} C(\theta(t)) & 0 \\ 0 & I \end{bmatrix}, \\ \bar{D}(\theta(t)) &= \begin{bmatrix} D(\theta(t)) & 0 \\ 0 & 0 \end{bmatrix}. \end{aligned} \quad (10)$$

Furthermore, $\bar{x}(t)^T = [x(t), x_c(t)] \in \mathbb{R}^{n+n_k}$ is the augmented state vector, $\bar{y}(t)^T = [y(t), x_c(t)] \in \mathbb{R}^{l+n_k}$ is the augmented output vector, and $\bar{u}(t)^T = [u(t), x_c(t)] \in \mathbb{R}^{m+n_k}$ is the augmented input vector. The augmented feedback gain is then:

$$\bar{F}(\theta(t)) = \begin{bmatrix} D_c(\theta(t)) & C_c(\theta(t)) \\ B_c(\theta(t)) & A_c(\theta(t)) \end{bmatrix}. \quad (11)$$

To obtain a PID controller design in the form:

- 1) PID controller with filter on the derivative part with filter time constant T_{fd} :

$$\begin{aligned} u(t) &= K_p(\theta(t))y(t) + K_i(\theta(t)) \int_0^\infty y(t)dt \\ &+ K_d(\theta(t)) \frac{1}{T_{fd}s + 1} \dot{y}(t), \end{aligned} \quad (12)$$

- 2) PID controller with filtered input with filter time constant T_f :

$$\begin{aligned} u(t) &= \frac{1}{T_f s + 1} \left(K_p(\theta(t))y(t) \right. \\ &\left. + K_i(\theta(t)) \int_0^\infty y(t)dt + K_d(\theta(t))\dot{y}(t) \right), \end{aligned} \quad (13)$$

one can prescribe the structure of the dynamic output-feedback controller matrices form (9) as follows:

$$\begin{aligned} A_c &= \begin{bmatrix} A_{c11} & \dots & A_{c1m} \\ \vdots & \ddots & \vdots \\ A_{cm1} & \dots & A_{cmm} \end{bmatrix}, B_c = \begin{bmatrix} B_{c11} & \dots & B_{c1l} \\ \vdots & \ddots & \vdots \\ B_{cm1} & \dots & B_{cml} \end{bmatrix}, \\ C_c &= \begin{bmatrix} C_{c11} & \dots & C_{c1m} \\ \vdots & \ddots & \vdots \\ C_{cm1} & \dots & C_{cmm} \end{bmatrix}, D_c = \begin{bmatrix} D_{c11} & \dots & D_{c1l} \\ \vdots & \ddots & \vdots \\ D_{cm1} & \dots & D_{cml} \end{bmatrix}, \end{aligned} \quad (14)$$

where the matrices $A_{c_{ij}} \in \mathbb{R}^{2 \times 2}$, $B_{c_{ij}} \in \mathbb{R}^{2 \times 1}$, $C_{c_{ij}} \in \mathbb{R}^{1 \times 2}$, and $D_{c_{ij}} \in \mathbb{R}^{1 \times 1}$ are:

$$\begin{aligned} A_{c_{ij}} &= \begin{bmatrix} -a_1 & 1 \\ -a_2 & 0 \end{bmatrix}, B_{c_{ij}} = \begin{bmatrix} b_{1ij}(\theta(t)) - a_1 b_{0ij}(\theta(t)) \\ b_{2ij}(\theta(t)) - a_2 b_{0ij}(\theta(t)) \end{bmatrix} \\ C_{c_{ij}} &= [1, 0], D_{c_{ij}} = [b_{0ij}(\theta(t))], \end{aligned} \quad (15)$$

where

1) For the PID controller with filter on the derivative part:

$$\begin{aligned} b_{0ij}(\theta(t)) &= \frac{k_{p_{ij}}(\theta(t))T_{fd} + k_{d_{ij}}(\theta(t))}{T_{fd}}, \\ b_{1ij}(\theta(t)) &= \frac{k_{p_{ij}}(\theta(t)) + k_{i_{ij}}(\theta(t))T_{fd}}{T_{fd}}, \\ b_{2ij}(\theta(t)) &= \frac{k_{i_{ij}}(\theta(t))}{T_{fd}}, a_1 = \frac{1}{T_{fd}}, a_2 = 0. \end{aligned}$$

2) For the PID controller with filtered input:

$$\begin{aligned} b_{0ij}(\theta(t)) &= \frac{k_{d_{ij}}(\theta(t))}{T_f}, b_{1ij}(\theta(t)) = \frac{k_{p_{ij}}(\theta(t))}{T_f}, \\ b_{2ij}(\theta(t)) &= \frac{k_{i_{ij}}(\theta(t))}{T_f}, a_1 = \frac{1}{T_f}, a_2 = 0. \end{aligned}$$

The controller gain matrices are affinely dependent on the scheduling variable as (3) with $S(\theta(t)) = \{K_p(\theta(t)), K_i(\theta(t)), K_d(\theta(t))\} \in \mathbb{R}^{m \times l}$. For centralized controller design the gain matrices $K_p(\theta(t))$, $K_i(\theta(t))$ and $K_d(\theta(t))$ are full matrices:

$$K_x(\theta(t)) = \begin{bmatrix} k_{x11}(\theta(t)) & \dots & k_{x1l}(\theta(t)) \\ \vdots & \ddots & \vdots \\ k_{xm1}(\theta(t)) & \dots & k_{xml}(\theta(t)) \end{bmatrix}, x = p, i, d. \quad (16)$$

To obtain decentralized control, the structure of these matrices can be predefined. In the case when $m = l$ a fully decentralized control can be obtained by structuring the gain matrices to diagonal form.

The main goal of this paper is to design a controller (in the form (5)) which guarantees the closed-loop stability and minimizes the cost function defined as:

$$J_\infty = \int_0^\infty J(\theta(t)) dt = \int_0^\infty \tilde{z}^T Q(\theta(t)) \tilde{z} dt \quad (17)$$

where $\tilde{z}^T = [\dot{x}(t), x(t), u(t), y(t)] \in \mathbb{R}^{2n+m+l}$ and

$$Q(\theta(t)) = \begin{bmatrix} Q_{\dot{x}}(\theta), N_{\dot{x}x}(\theta), N_{\dot{x}u}(\theta), N_{\dot{x}y}(\theta) \\ *, Q_x(\theta), N_{xu}(\theta), N_{xy}(\theta) \\ *, *, R(\theta), N_{uy}(\theta) \\ *, *, *, Q_y(\theta) \end{bmatrix} \geq 0.$$

The weighting matrix $Q(\theta(t))$ is affinely dependent on the scheduling variable defined as (3) with $S(\theta(t)) = \{Q(\theta(t))\}$. In addition, $Q_{\dot{x}}(\theta(t)) \in \mathbb{R}^{n \times n} \geq 0$, $Q_x(\theta(t)) \in \mathbb{R}^{n \times n}$, $Q_y(\theta(t)) \in \mathbb{R}^{l \times l}$, $R(\theta(t)) \in \mathbb{R}^{m \times m} > 0$ and matrices $N_{ij}(\theta(t))$, $i, j = \dot{x}, x, u, y$, $i \neq j$ are cross-term weighting matrices with appropriate dimensions.

Remark 1. Controllability and observability analysis for LPV systems can be done by using the approach presented in [26].

III. ROBUST LPV BASED LQR DESIGN

The main result of this paper is formulated in the next theorem. For the shake of simplicity, all the undefined expressions will be explained in the proof of the next theorem.

Theorem 1. For the uncertain affine LPV system (2) an optimal (suboptimal) LQR-based controller exists in the form (5), if for the given weighting matrix $Q(\theta(t))$ and predefined maximal rate of change of scheduled parameters ρ_θ , the following conditions hold:

$$\min_{F, P, X} \text{trace} \left(\sum_{i=1}^{2^p} \tilde{P}_i \right) \quad (18)$$

subject to:

$$\tilde{M}_i = \begin{bmatrix} \tilde{M}_{i11} & \dots & \tilde{M}_{i41} \\ \vdots & \ddots & \vdots \\ \tilde{M}_{i41} & \dots & \tilde{M}_{i44} \end{bmatrix} \leq 0, \quad (19)$$

$$\tilde{P}_i > 0, \quad (20)$$

$$i = 1, 2, \dots, 2^p,$$

where

$$\begin{aligned} \tilde{M}_{i11} &= \tilde{E}_i^T X_1 + X_1^T \tilde{E}_i + \tilde{S}_i, \\ \tilde{M}_{i21} &= -\tilde{A}_i^T X_1 + X_2^T \tilde{E}_i - \tilde{C}^T X_9 + \tilde{N}_{i\dot{x}x}^T, \\ \tilde{M}_{i31} &= -\tilde{B}_i^T X_1 + X_3^T \tilde{E}_i + X_5 - \tilde{D}_i^T X_9 + \tilde{N}_{i\dot{x}u}^T, \\ \tilde{M}_{i41} &= X_4^T \tilde{E}_i - \tilde{F}_i^T X_5 + X_9 + \tilde{N}_{i\dot{x}y}^T, \\ \tilde{M}_{i22} &= -X_2^T \tilde{A}_i - \tilde{A}_i^T X_2 - X_{10}^T \tilde{C}_i - \tilde{C}_i^T X_{10} + \tilde{Q}_{i\dot{x}}, \\ \tilde{M}_{i32} &= -\tilde{B}_i^T X_2 - X_3^T \tilde{A}_i + X_6 - \tilde{D}_i^T X_{10} \\ &\quad - X_{11}^T \tilde{C}_i + \tilde{N}_{i\dot{x}u}^T, \\ \tilde{M}_{i42} &= -X_4^T \tilde{A}_i - \tilde{F}_i^T X_6 + X_{10} - X_{12}^T \tilde{C}_i + \tilde{N}_{i\dot{x}y}^T, \\ \tilde{M}_{i33} &= -X_3^T \tilde{B}_i - \tilde{B}_i^T X_3 + X_7^T + X_7 - X_{11}^T \tilde{D}_i \\ &\quad - \tilde{D}_i^T X_{11} + \tilde{R}_i, \\ \tilde{M}_{i43} &= -X_4^T \tilde{B}_i - \tilde{F}_i^T X_7 + X_8^T + X_{11} \\ &\quad - X_{12}^T \tilde{D}_i + \tilde{N}_{iuy}^T, \\ \tilde{M}_{i44} &= -X_8^T \tilde{F}_i - \tilde{F}_i^T X_8 + X_{12}^T + X_{12} + \tilde{Q}_{iy}. \end{aligned}$$

Proof. The proof is based on the free matrix weighting approach:

$$2(\dot{x}^T X_1^T + x^T X_2^T + u^T X_3^T + y^T X_4^T) (\dot{x} - A(\theta)x - B(\theta)u) = 0, \quad (21)$$

$$2(\dot{x}^T X_5^T + x^T X_6^T + u^T X_7^T + y^T X_8^T) (y - C(\theta)x - D(\theta)u) = 0, \quad (22)$$

$$2(\dot{x}^T X_9^T + x^T X_{10}^T + u^T X_{11}^T + y^T X_{12}^T) (u - F(\theta)y) = 0. \quad (23)$$

The matrices X_i , $i = 1, \dots, 12$ are auxiliary matrices with appropriate dimensions. One can transform the equalities (21)-(23) to the form:

$$\tilde{z}^T(t)W(\theta)\tilde{z}(t) = 0 \quad (24)$$

where

$$\begin{aligned} W_{11} &= E(\theta)^T X_1 + X_1^T E(\theta), \\ W_{21} &= -A(\theta)^T X_1 + X_2^T E(\theta) - C(\theta)^T X_9, \\ W_{31} &= -B(\theta)^T X_1 + X_3^T E(\theta) + X_5 - D(\theta)^T X_9, \\ W_{41} &= X_4^T E(\theta) - F(\theta)^T X_5 + X_9, \\ W_{22} &= -X_2^T A(\theta) - A(\theta)^T X_2 - X_{10}^T C(\theta) \\ &\quad - C(\theta)^T X_{10}, \\ W_{32} &= -B(\theta)^T X_2 - X_3^T A(\theta) + X_6 - D(\theta)^T X_{10} \\ &\quad - X_{11}^T C(\theta), \\ W_{42} &= -X_4^T A(\theta) - F(\theta)^T X_6 + X_{10} - X_{12}^T C(\theta), \\ W_{33} &= -X_3^T B(\theta) - B(\theta)^T X_3 + X_7^T + X_7 \\ &\quad - X_{11}^T D(\theta) - D(\theta)^T X_{11}, \\ W_{43} &= -X_4^T B(\theta) - F(\theta)^T X_7 + X_8^T + X_{11} \\ &\quad - X_{12}^T D(\theta), \\ W_{44} &= -X_8^T F(\theta) - F(\theta)^T X_8 + X_{12}^T + X_{12}. \end{aligned}$$

Let's choose the Lyapunov function as:

$$V(\theta(t)) = x(t)^T P(\theta(t))x(t), \quad (25)$$

where

$$P(\theta(t)) = P_0 + \sum_{i=1}^p P_i \theta_i(t). \quad (26)$$

The first derivative of the Lyapunov function (25) is then:

$$\begin{aligned} \dot{V}(\theta(t)) &= \dot{x}^T(t)P(\theta(t))x(t) + x^T(t)P(\theta(t))\dot{x}(t) \\ &\quad + x^T(t)P(\dot{\theta}(t))x(t). \end{aligned} \quad (27)$$

where

$$P(\dot{\theta}(t)) \leq \sum_{i=1}^p P_i \rho_i = P_\rho. \quad (28)$$

One can transform the first derivative of the Lyapunov function (27) with (28) to the form:

$$\dot{V}(\theta(t)) = \tilde{z}^T(t) \begin{bmatrix} 0, & P(\theta(t)), & 0, & 0 \\ P(\theta(t)), & P_\rho, & 0, & 0 \\ 0, & 0, & 0, & 0 \\ 0, & 0, & 0, & 0 \end{bmatrix} \tilde{z}(t). \quad (29)$$

By summarizing the equations (17), (24) and (29), the Bellman-Lyapunov inequality can be obtained in the form:

$$\begin{aligned} \tilde{z}^T(t)(W(\theta(t)) + \dot{V}(\theta(t)) + J(\theta(t)))\tilde{z}(t) \\ = \tilde{z}^T(t)M(\theta(t))\tilde{z}(t) \leq 0, \end{aligned} \quad (30)$$

Furthermore, if $P(\theta(t))$ is positive definite then the Bellman-Lyapunov inequality (30) can be rewritten to this form ($W(\theta(t)) = 0$):

$$\dot{V}(\theta(t)) + J(\theta(t)) \leq 0 \rightarrow \dot{V}(\theta(t)) \leq -J(\theta(t)) \leq 0 \quad (31)$$

Integrating both side from 0 to ∞ one can obtain:

$$J_\infty = \int_0^\infty J(\theta(t)) \leq V(0) - V(\infty) = x_0^T P(\theta(0))x_0. \quad (32)$$

It follows that by minimizing $x_0^T P(\theta(0))x_0$ and by satisfying $M(\theta(t)) \leq 0$ as well as $P(\theta(t)) > 0$ for all $\theta \in \Omega$, the closed-loop system will be affinely quadratically stable with guaranteed cost defined by (32).

Since $M(\theta(t))$ and $P(\theta(t))$ are convex regarding the scheduling variable θ , therefore $M(\theta(t))$ will be negative semi-definite and $P(\theta(t))$ will be positive definite if and only if they are negative semi-definite and definite at the corners of θ . Hence, definiteness/semi-definiteness splits to 2×2^p inequalities \rightarrow (19) and (20). The tilde denotes the given elements at the vertices of θ .

If the initial condition of the scheduling variable is available then it is enough to minimize $\text{trace}(P(\theta(0)))$. Otherwise we recommend to minimize: $\text{trace}\left(\sum_{i=1}^{2^p} \tilde{P}_i\right)$. \square

IV. EXAMPLES

In order to show the viability of the previous proposed method, the following two examples have been chosen.

Example 1. The first example is a simple multi-input multi-output (MIMO) linear time-invariant (LTI) system, which will be used to demonstrate and compare the proposed method with the standard LQR design. The transfer function of the system is as follows:

$$G(s) = \begin{bmatrix} \frac{2}{10s+1}, & \frac{2}{s+1} \\ \frac{1}{s+1}, & \frac{-4}{10s+1} \end{bmatrix}, \quad (33)$$

which can be transformed to the form (2):

$$A = \begin{bmatrix} -0.1, & 0, & 0, & 0 \\ 0, & -1, & 0, & 0 \\ 0, & 0, & -1, & 0 \\ 0, & 0, & 0, & -0.1 \end{bmatrix}, B = \begin{bmatrix} 0.5, & 0 \\ 1, & 0 \\ 0, & 2 \\ 0, & 0.5 \end{bmatrix},$$

$$C = \begin{bmatrix} 0.4, & 0, & 1, & 0 \\ 0, & 1, & 0, & -0.8 \end{bmatrix}, D = \begin{bmatrix} 0, & 0 \\ 0, & 0 \end{bmatrix}, E = \begin{bmatrix} 1, & 0, & 0, & 0 \\ 0, & 1, & 0, & 0 \\ 0, & 0, & 1, & 0 \\ 0, & 0, & 0, & 1 \end{bmatrix}.$$

Different controller types were designed: static state feedback (SSF), centralized/decentralized SOF, reduced/full order dynamic state feedback (DSF), reduced/full order DOF, centralized/decentralized PI and centralized/decentralized PID. Numerical solution has been carried out by PENBMI 2.1 [27] / PENLAB 1.04 [28] solver under MATLAB 2014b using YALMIP R20150918 [29]. The simulations were done via SIMULINK. The obtained guaranteed cost (J_∞) for:

$$Q_x = \begin{bmatrix} 1, & 0, & 0, & 0 \\ 0, & 1, & 0, & 0 \\ 0, & 0, & 1, & 0 \\ 0, & 0, & 0, & 1 \end{bmatrix}, R = \begin{bmatrix} 1, & 0 \\ 0, & 1 \end{bmatrix}, N_{xu} = \begin{bmatrix} 0.1, & 0.1 \\ 0.1, & 0.1 \\ 0.1, & 0.1 \\ 0.1, & 0.1 \end{bmatrix}, \quad (34)$$

$$Q_y, Q_{\dot{x}}, N_{i,j} = 0, \quad i, j = \dot{x}, x, y, u; \quad i \neq j; \quad i, j \neq x, u$$

can be found in Table 1. Similar comparison with output weighting can be found in Table 2 for:

$$Q_y = \begin{bmatrix} 1, & 0 \\ 0, & 1 \end{bmatrix}, R = \begin{bmatrix} 1, & 0 \\ 0, & 1 \end{bmatrix}, N_{uy} = \begin{bmatrix} 0.1, & 0.1 \\ 0.1, & 0.1 \end{bmatrix}, \quad (35)$$

$$Q_x, Q_{\dot{x}}, N_{i,j} = 0, \quad i, j = \dot{x}, x, y, u; \quad i \neq j; \quad i, j \neq u, y$$

Example 2. The second example is an arm-driven inverted pendulum (ADIP) with strong nonlinearity described in [30]

TABLE I
CONTROLLER TYPES & GUARANTEED COSTS

Controller type	J_∞
<i>Standard infinite-horizon LQR:</i>	
▷ SSF	3.6913
▷ DSF	3.6749
<i>Proposed method:</i>	
▷ SSF	3.6913
▷ Centralized SOF	4.9736
▷ Decentralized SOF	5.8906
▷ DSF ($n_k = 4$)	3.6749
▷ DOF ($n_k = 2$)	4.9699
▷ Centralized PI	9.9376
▷ Decentralized PI	13.2005
▷ Decentralized PID ₁ ($T_{fd} = 0.5$)	11.3854
▷ Decentralized PID ₂ ($T_f = 0.5$)	11.3854

TABLE II
CONTROLLER TYPES & GUARANTEED COSTS

Controller type	J_∞
<i>Standard infinite-horizon LQR with output weighting:</i>	
▷ SSF	1.0013
▷ DSF	0.9909
<i>Proposed method:</i>	
▷ SSF	1.0013
▷ Centralized SOF	1.1002
▷ Decentralized SOF	2.4490
▷ DSF ($n_k = 2$)	0.9962
▷ DOF ($n_k = 2$)	1.0026

(Fig. 1). The ADIP has two links with joints and encoders to measure the relative angles. The first joint is equipped with a geared DC motor. The main goal is to keep the second link in upright position for a given reference angle of the first joint. Furthermore, to ensure $|u| \leq 0.35$ [p.m] while the max step size for φ_1 can be 60 [deg].

The nonlinear model of the ADIP can be described with the following motion equation:

$$\begin{bmatrix} \alpha_1 & \alpha_3 \cos \varphi_{12} \\ \alpha_3 \cos \varphi_{12} & \alpha_2 \end{bmatrix} \begin{bmatrix} \ddot{\varphi}_1 \\ \ddot{\varphi}_2 \end{bmatrix} + \begin{bmatrix} \alpha_3 \dot{\varphi}_2^2 \sin \varphi_{12} \\ -\alpha_3 \dot{\varphi}_1^2 \sin \varphi_{12} \end{bmatrix} + \begin{bmatrix} -\alpha_4 \sin \varphi_1 \\ -\alpha_5 \sin \varphi_2 \end{bmatrix} + \begin{bmatrix} c_1 + c_2 & -c_2 \\ -c_2 & c_2 \end{bmatrix} \begin{bmatrix} \dot{\varphi}_1 \\ \dot{\varphi}_2 \end{bmatrix} = \begin{bmatrix} \tau \\ 0 \end{bmatrix} \quad (36)$$

where $\varphi_{12} = \varphi_1 - \varphi_2$, $\alpha_1 = J_{g1} + m_1 l_1^2 + m_2 L_1^2$, $\alpha_2 = J_{g2} + m_2 l_2^2$, $\alpha_3 = m_2 L_1 l_2$, $\alpha_4 = m_1 g l_1 + m_2 g L_1$, $\alpha_5 = m_2 g l_2$, furthermore, m_i is the i -th link's mass, c_i is the viscous friction coefficient of the joint i , J_{gi} is the moment of inertia about the center of gravity of the link i , l_i is the length between the center of gravity of the link i and the joint i , L_1 is the length between the first and the second joint, and g is the gravity acceleration. The physical parameters are shown in Table 3.

Since the active joint is operated by a geared DC motor and a velocity control driver, the model (36) can be rewritten to:

$$\begin{bmatrix} 1 & 0 \\ \alpha_3 \cos \varphi_{12} & \alpha_2 \end{bmatrix} \begin{bmatrix} \ddot{\varphi}_1 \\ \ddot{\varphi}_2 \end{bmatrix} + \begin{bmatrix} 0 \\ -\alpha_3 \dot{\varphi}_1^2 \sin \varphi_{12} \end{bmatrix} + \begin{bmatrix} 0 \\ -\alpha_5 \sin \varphi_2 \end{bmatrix} + \begin{bmatrix} a_s & 0 \\ -c_2 & c_2 \end{bmatrix} \begin{bmatrix} \dot{\varphi}_1 \\ \dot{\varphi}_2 \end{bmatrix} = \begin{bmatrix} b_s \\ 0 \end{bmatrix} u \quad (37)$$

where u is the control input and $a_s = 6.25$, $b_s = 15.6$.

TABLE III
SYSTEM PARAMETERS

$m_1 :$	3.06×10^{-1} [kg]	$m_2 :$	1.04×10^{-1} [kg]
$l_1 :$	1.20×10^{-1} [m]	$l_2 :$	1.75×10^{-1} [m]
$L_1 :$	2.27×10^{-1} [m]	$c_2 :$	1.86×10^{-4} [kg.m ² /s]
$J_{g1} :$	1.38×10^{-3} [kg.m ²]	$J_{g2} :$	1.38×10^{-4} [kg.m ²]

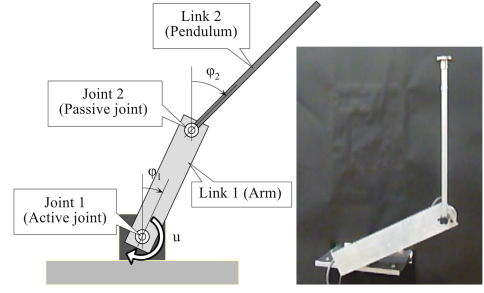


Fig. 1. Arm-driven inverted pendulum [30].

The model (37) can be rewritten to the form (2), with the assumption that $\sin \varphi_2 = \varphi_2$ (since we assume that $\varphi_2 \leq 2$ [deg]), with $x = [\varphi_1, \varphi_2, \omega_1, \omega_2]^T$ and with scheduled parameters (with assumption that $\varphi_1 \leq 65$ [deg]) defined as $\theta_1 = 1 - \cos(x_1 - x_2) \in \langle 0, 0.557 \rangle$, $\theta_2 = -\alpha_3 x_3 \sin(x_1 - x_2) \in \langle -0.003, 0.003 \rangle$, $\dot{\theta}_1 \leq 0.4$ and $\dot{\theta}_2 \leq 6 \times 10^{-3}$.

$$A(\theta) = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & 0 & -a_s & 0 \\ 0 & \alpha_5 & p_2(\theta) & -c_1 \end{bmatrix}, B = \begin{bmatrix} 0 \\ 0 \\ b_s \\ 0 \end{bmatrix},$$

$$C = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}, D = \begin{bmatrix} 0 & 0 \\ 0 & 0 \\ 0 & 0 \\ 0 & 0 \end{bmatrix}, E(\theta) = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & p_1(\theta) & \alpha_2 \end{bmatrix},$$

$$p_1(\theta) = \alpha_3 - \alpha_3 \theta_1, p_2(\theta) = c_2 + \theta_2.$$

The obtained model was extended for PI controller design as (7) with integral part for φ_1 . Then a gain-scheduled PI controller was designed in the form (5) with weighting matrices:

$$\begin{aligned} Q_x(\theta) &= \text{diag}([q_1(\theta), 1, 0.1, 0.1, q_5(\theta)]), R = 1, \\ q_1(\theta) &= 2.8 + 2\theta_1, q_5(\theta) = 18 - 10\theta_2, \\ Q_y, Q_{\dot{x}}, N_{i,j} &= 0, i, j = \dot{x}, x, y, u; i \neq j. \end{aligned} \quad (38)$$

The obtained controller gain matrices are as follows:

$$\begin{aligned} F_0 &= [12.1179, 106.1869, 7.0281, 14.4041, 8.8046], \\ F_1 &= [-13.0849, -49.2797, -5.9001, -6.6199, -9.8506], \\ F_2 &= 1 \times 10^3 [0.3627, 2.8745, 0.3010, 0.3908, 0.2484]. \end{aligned}$$

Numerical solution has been carried out by MOSEK LMI solver under MATLAB 2014b using YALMIP R20150918 [29] by reducing the BMI problem into iterative LMI (by fixing parameters related to controller matrices). The simulations (Fig. 2) were done via SIMULINK using the nonlinear model (36).

V. CONCLUSION

A novel approach for optimal/suboptimal LPV-based robust LQR design is presented in this paper for uncertain affine LPV systems with inexact scheduled parameters. The proposed design method extends the standard infinite-horizon LQR design to linear parameter-varying SOF, DOF, PI and PID controller design. Moreover, the obtained stability and performance requirements are convex functions of the scheduled parameters without any restriction on the controller and system

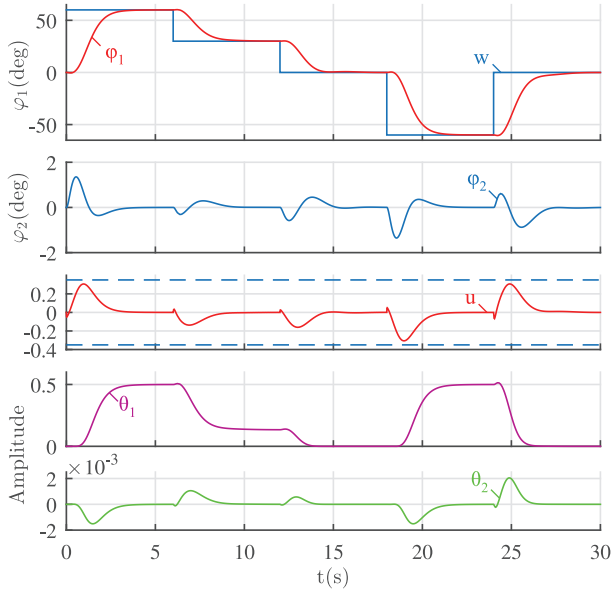


Fig. 2. Simulation results using the nonlinear model (36).

matrices. Therefore, the proposed approach delivers a less conservative and more usable design approach.

APPENDIX A

Inexact scheduled parameters

Assume that the measurable/estimable scheduling variable $\bar{\alpha}_i(t)$ consists of an exact part $\alpha_i(t)$ and from an error (inexact) part $\delta_i(t)$:

$$\bar{\alpha}_i(t) = \alpha_i(t) + \delta_i(t), \quad i = 1, 2, \dots, N_\alpha. \quad (39)$$

For the affine form we get:

$$\begin{aligned} S(\bar{\alpha}(t)) &= S_0 + \sum_{i=1}^{N_\alpha} S_i(\alpha_i(t) + \delta_i(t)) \\ &= S_0 + \sum_{i=1}^{N_\alpha} S_i\alpha_i(t) + \sum_{i=1}^{N_\alpha} S_i\delta_i(t). \end{aligned} \quad (40)$$

If only some of the scheduling variables has non-zero error part than (40) can be rewritten as:

$$S(\bar{\alpha}(t)) = S_0 + \sum_{i=1}^{N_\alpha} S_i\alpha_i(t) + \sum_{j=1}^{N_\delta} S_{(i+j)}\delta_j(t), \quad (41)$$

where $S_{(i+j)}$ are the matrices for which the error part is non-zero.

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