

Robust adaptive control of switched-reluctance motors without velocity measurements

Erik Chumacero Antonio Loría Gerardo Espinosa-Pérez

Abstract— We present a speed-sensorless tracking controller for switched reluctance motors with unknown parameters. Our approach relies on the design of two control loops: an outer control-loop for the rotor dynamics which is driven by a PID-type controller where the stator currents are viewed as virtual control inputs, and an inner tracking control-loop for the stator currents. We assume that the parameters of the rotor (inertia and the load torque) are unknown and we establish uniform global exponential stability. In the case that also the stator parameters are unknown, we add an adaptation law and we establish convergence of the tracking errors.

I. INTRODUCTION

Switched Reluctance Machines (SRM) are attractive since they are reliable, relatively cheap and they produce high torque at low speed, which makes them suitable for direct-drive applications. However, even under experimentally-validated assumptions leading to some simplifications, the dynamic model is highly nonlinear *e.g.*, the generated electromagnetical torque is a quadratic function of the electric currents and rotor positions. In addition, reliable and accurate indirect sensing methods for the mechanical variables are fundamental in the development of low-cost, high-performance SRM drives; on one hand the use of mechanical sensors increase the cost of the set-up and on the other, velocity sensors are often contaminated with noise. Therefore, avoiding the use of angular velocity and position sensors which is well known as *sensorless* control, is beyond pure theoretical interest.

There exist a large number of efficient heuristically-based and experimentally-validated control approaches to reduce the number of mechanical sensors in the loop –see *e.g.*, [1], [2], [3], [4], [5]. However, articles on control of switched-reluctance drives that include a rigorous stability analysis, especially in a sensorless context, are rare. The main result in [6] establishes global asymptotic stability for a passivity-based controller in the case of unknown load however, it uses both¹ mechanical variables. A proportional-derivative-based controller is proposed in [7] but relying on the knowledge of the torque load. In [8] the authors consider only the rotor dynamics that is, it is assumed that the currents are valid physical control inputs. An adaptive position-feedback controller is presented. The controller uses a dirty derivatives filter instead of velocity measurements and it is guaranteed

that the tracking errors converge to zero. An interesting experimental validation is presented as well to compare the performance against a full-state feedback controller.

In this paper we establish uniform global exponential stability for the SRM under position-feedback control. We use the full model, including both the rotor and stator dynamics and we assume that the rotor inertia and the load torque are unknown. Then, under the assumptions that the rest of the physical parameters are unknown we establish uniform global stability and the convergence of the tracking errors to zero. Our control approach relies on a minimal parameterization, a tracking controller for the stator dynamics and the so-called PI²D controller, introduced in [9], for the rotor dynamics. This controller is of a PID type except that the velocity measurements are replaced by dirty derivatives.

The rest of the paper is organized as follows. In next section we briefly describe the motor model, in Section III we discuss the rationale behind our main results, presented in Section IV. In Section V we show some simulation results and we conclude with a few remarks in Section VI.

II. THE MOTOR MODEL

After experimental evidence, it is well accepted that the three stator phases of a switched-reluctance motor may be assumed to be magnetically decoupled *i.e.*, the mutual inductance among stator phases is negligible [10]. Furthermore, based on the assumption that the machine operates at relatively low current levels, it is common practice to express the inductance of each phase as a strictly positive Fourier series truncated at the first harmonic that is,

$$L_j(q) = \ell_0 - \ell_1 c_j, \quad c_j := \cos\left(N_r q - (j-1)\frac{2\pi}{m}\right)$$

where q denotes the rotor angular position, $\ell_0 > \ell_1 > 0$ are inductance values and N_r is the number of poles. Under these assumptions the dynamic model for the stator currents is given by

$$u_j = L_j(q)\dot{x}_j + K_j(q)\omega x_j + R x_j \quad (1)$$

where the index $j \in \{1, \dots, m\}$ with m is the number of phases, x_j denotes currents,

$$K_j(q) = \frac{\partial L_j}{\partial q} = N_r \ell_1 s_j, \quad s_j := \sin\left(N_r q - (j-1)\frac{2\pi}{m}\right)$$

corresponds to the phase-inductance variation, u_j is the input voltage at the stator terminals (and control input) and R is the stator winding resistance. The rotor dynamics is given by a simple integrator of the input torques, these are the load torque τ_L and the mechanical torque of electromechanical

A. Loría is with CNRS. E. Chumacero is with Univ. Paris Sud. Address: LSS-SUPELEC, 91192 Gif-sur-Yvette, France. E-mail: antonio.loria@lss.supelec.fr, erik.chumacero@lss.supelec.fr

G. Espinosa-Pérez is with DEPEFI – UNAM, A.P. 70-256, 04510 México D.F., MEXICO. gerardoe@servidor.unam.mx.

¹Angular velocity and position measurements

origin, which based on the assumption that the behavior of the stator windings is decoupled, is given by

$$\tau_e(q, x) = \sum_{j=1}^m K_j(q)x_j^2.$$

Thus the complete motor model is given by

$$L(q)\dot{x} + K(q)\omega x + Rx = u \quad (2a)$$

$$J\dot{\omega} = \tau_e(q, x) - \tau_L(q, \omega) \quad (2b)$$

where $L := \text{diag}\{L_j\}$, $K := \text{diag}\{K_j\}$, $R := \text{diag}\{R_j\}$ and $x := \text{col}[x_1, x_2, x_3]^\top$; for further development, we remark that there exist constants ℓ_m , ℓ_M and k_M such that for all $q \in [-\pi, \pi]$,

$$0 < \ell_m \leq |L_j(q)| \leq \ell_M, \quad |K_j(q)| \leq k_M. \quad (3)$$

Equation (2b) models the rotor dynamics; J denotes the rotor inertia and ω denotes the angular velocity that is, $\omega = \dot{q}$. In spite of the simplifications made for control analysis purposes, this model is adopted in both the electrical-machines and the control research communities –cf. [11].

We consider a three-phase electric motor that is, $m = 3$. The control goal is to design a dynamic controller for (2) whose output $u = [u_1 \ u_2 \ u_3]^\top$ depending on the stator currents and rotor angular positions, such that $\omega(t)$ tracks bounded smooth reference trajectories ω^* . We make the standing assumption that τ_L and J are both unknown and $J \in [J_m, J_M]$ with known limits J_m and J_M . Furthermore $\dot{\omega}^*$ is assumed (piecewise) constant.

III. CONTROL STRATEGY

The control approach consists in applying two control loops; an outer loop to stabilize the rotor dynamics via angular position measurement and an inner loop to stabilize the stator dynamics using currents measurement. More precisely, given desired reference ω^* we design a *desired* control input τ_d for the mechanical equation (2b), such that $\tau_e^* = \tau_d$ implies that $\omega \rightarrow \omega^*$. To that end, we define a *reference* mechanical torque

$$\tau_e^*(q, x^*) = \frac{1}{2} \left(K_1(q)x_1^{*2} + K_2(q)x_2^{*2} + K_3(q)x_3^{*2} \right) \quad (4)$$

where x_j^* is a current reference trajectory for each phase and which is defined as a solution to

$$\tau_e^* = \tau_d \quad (5)$$

for any given τ_d . That is, the stator control loop is to ensure that $x \rightarrow x^*$ so that $\tau_e \rightarrow \tau_d$ and in turn, $\omega \rightarrow \omega^*$. Next, we explain in detail both controllers for the rotor and the stator dynamics.

A. Control of rotor velocity

The complexity of the rotor dynamics equation resides in the fact that it is non-affine in the (virtual control) inputs x ; such difficulty is overcome by using the torque-sharing technique –see [7], explained farther below. For the time being, let us consider that τ_e is a virtual control input to (2b) then, (5) implies that

$$J\dot{\omega} = \tau_d - \tau_L + \tau_e - \tau_e^* \quad (6)$$

For the purpose of designing the control law τ_d for (6) we see the latter as a simple integrator of the unknown constant τ_L and we consider $\tau_e - \tau_e^*$ as a vanishing perturbation. The choice of proportional-integral-derivative control (PID) comes naturally; furthermore, since ω is assumed to be unmeasurable, we use the PI²D controller, introduced in [9] for robot manipulators. It corresponds to a modified PID controller in which the ‘derivative’ term is proportional to a filtered velocity vector and double integral action, both on e_q and ϑ , is used. The fact that PID control is model-free is particularly suitable in the context of parametric uncertainty. The PI²D controller, for the rotor dynamics is defined by

$$\tau_d = -k_p e_q - k_d \vartheta + \nu + \hat{J} \dot{\omega}^* \quad (7a)$$

$$\dot{\nu} = -k_i (e_q - \vartheta) \quad (7b)$$

$$\dot{q}_c = -a(q_c + b e_q) \quad (7c)$$

$$\vartheta = q_c + b e_q \quad (7d)$$

where k_p, k_i, k_d, a, b are positive reals, $\hat{J} \in [J_m, J_M]$ is a *constant* estimate of J and $e_q = q - q^*$ with

$$q^*(t) = \int_0^t q^*(s) ds, \quad q^*(0) = q_0^* \in [-\pi, \pi].$$

Note that τ_d is independent on ω . Since the variable to be controlled is ω , the initial value of q_0^* is innocuous. It is also important to remark that $\hat{J} \dot{\omega}^*$ is (piecewise) constant by assumption hence, its effect and that of the load torque τ_L may be compensated for by integral action, as a matter of fact ν converges (albeit slowly) to

$$\nu^* := \tau_L - \tilde{J} \dot{\omega}^* \quad (8)$$

where $\tilde{J} = \hat{J} - J$. The last two equations in (7) correspond to the well-known and widely used ‘dirty derivatives’. The nickname comes from the observation that it is equivalent to

$$\vartheta = \frac{b}{s+a} e_\omega$$

where s is the Laplace variable. Note that ϑ is *not* a converging estimate of the velocity e_ω that is $\vartheta \not\rightarrow e_\omega$, save in the limit case when the pole is placed at $-\infty$ and the DC gain $b/a = 1$. Thus, using (7a) in (6), $\hat{J} = \tilde{J} + J$ and defining

$$z := \nu + \tilde{J} \dot{\omega}^* - \frac{k_i}{\varepsilon} e_q - \tau_L, \quad k_p' = k_p - \frac{k_i}{\varepsilon}$$

we see that the *mechanical* equation becomes

$$\begin{bmatrix} \dot{e}_q \\ \dot{e}_\omega \\ \dot{\vartheta} \\ \dot{z} \end{bmatrix} = \underbrace{\begin{bmatrix} 0 & 1 & 0 & 0 \\ -k_p'/J & 0 & -k_d/J & 1/J \\ 0 & b & -a & 0 \\ -k_i & -k_i/\varepsilon & k_i & 0 \end{bmatrix}}_A \underbrace{\begin{bmatrix} e_q \\ e_\omega \\ \vartheta \\ z \end{bmatrix}}_{y_1} + \underbrace{\begin{bmatrix} 0 \\ 1/J \\ 0 \\ 0 \end{bmatrix}}_B u_\tau \quad (9)$$

where $u_\tau := \tau_e - \tau_e^*$. Note that we can make the matrix A Hurwitz by properly choosing the gains k_p, k_i, k_d, a and b without the knowledge of J . Consequently, the mapping $u_\tau \mapsto y_1$ is input to state stable. Indeed, the Hurwitz property

is equivalent to the existence of $P = P^\top > 0$ and $\gamma_1 > |PB|$ such that $Q = -(A^\top P + PA)$ is positive definite and

$$V_1(y_1) = \frac{1}{2} y_1^\top P y_1 \quad (10)$$

satisfies

$$\begin{aligned} \dot{V}_1(y_1) &= \frac{1}{2} y_1^\top (A^\top P + PA) y_1 + y_1^\top P B u_\tau \\ &\leq -\frac{1}{2} y_1^\top Q y_1 + \gamma_1 |y_1| |\tau_e - \tau_e^*|. \end{aligned} \quad (11)$$

The control loop of the stator dynamics, explained next, is responsible for making $|\tau_e - \tau_e^*| \rightarrow 0$.

B. Control of the stator dynamics

We stress that Equation (9) is equivalent to (2b) provided that (5) holds for τ_d defined as in (7a); to ensure that $\tau_e^*(q, x^*) = \tau_d$ we need to solve the latter for x^* . To that end, we employ the so-called torque sharing technique, introduced in [7] and which leads to the definition of a reference current trajectory x^* . According to (4) we have for each $j \in \{1, 2, 3\}$

$$x_j^* = \left[\frac{2\tau_d}{K_j(q)} \right]^{1/2}$$

however, for x_j^* to be well-posed $K_j(q)$ must be different from zero and its sign must be the same as that of τ_d . To ensure that this is the case, we exploit the physics of the reluctance machine as in [12], [7] and introduce a current-switching policy defined by commutation piecewise constant functions $q \mapsto m_j$ as follows. Let the sets

$$\begin{aligned} \Theta_j^+ &= \{q \in [-\pi, \pi] : s_j(q) \geq 0\} \\ \Theta_j^- &= \{q \in [-\pi, \pi] : s_j(q) < 0\} \end{aligned}$$

and let m_j be such that $\sum_{j=1}^3 m_j(q) = 1$,

$$m_j(q) = \begin{cases} m_j^+(q) & \text{if } \tau_d \geq 0, \\ m_j^-(q) & \text{if } \tau_d < 0. \end{cases}$$

where

$$\begin{aligned} m_j^+(q) &> 0 \quad \forall q \in \Theta^+, \quad m_j^+(q) = 0 \quad \forall q \in \Theta^-, \\ m_j^-(q) &> 0 \quad \forall q \in \Theta^-, \quad m_j^-(q) = 0 \quad \forall q \in \Theta^+. \end{aligned}$$

Because the functions s_j are sinusoids out of phase by $2\pi/3$, for each q and τ_d there always exists (at least) one $j \in \{1, 2, 3\}$ such that

$$\frac{2\tau_d m_j(q)}{K_j(q)} \in (0, \infty).$$

Furthermore, to smoothen the phase transitions one may use smooth functions m_j as opposed to piecewise constant and we introduce hysteresis around the switching condition $s_j = 0$ that is, let

$$x_j^*(\sigma) = \begin{cases} \sigma \left[\frac{\tau_d m_j(q)}{s_j(q)} \right]^{1/2} & \text{if } |s_j(q)| > \delta_K \\ 0 & \text{otherwise.} \end{cases} \quad (12)$$

where δ_K is the hysteresis design parameter and

$$\sigma = \sqrt{\frac{2}{N_r \ell_1}}. \quad (13)$$

Under these conditions, $\tau_e^*(q, x^*) = \tau_d$.

The rationale to solve the tracking control problem $x_j \rightarrow x_j^*$ builds upon the observation that under the action of the tracking control law

$$u = L(q)\dot{x}^* + \omega^* K(q)x + Rx - k_{px} e_x, \quad k_{px} > 0 \quad (14)$$

where $e_x := x - x^*$, the origin of the closed-loop equation

$$L(q)\dot{e}_x + k_{px} e_x = -K(q)x e_\omega, \quad e_\omega := \omega - \omega^* \quad (15)$$

is globally exponentially stable provided that $e_\omega \equiv 0$. Indeed, the system is reminiscent of a perturbed linear system with stable drift; to see this, note that $L(q)$ is positive definite and bounded uniformly in q therefore, exponential stability of the origin of

$$L(q)\dot{e}_x = -[R + k_{px}] e_x$$

is equivalent to that of the origin of

$$\dot{e}_x = -[R + k_{px}] e_x$$

which holds for any positive value of the control gain k_{px} . Furthermore, the system may be rendered input to state stable from the input e_ω provided that the gain k_{px} dominates over the input ‘‘gain’’ function $K(q)x$; note that this is feasible as q and x are measured states.

Although simple and appealing, the control law (14) is not implementable since \dot{x}^* depends on the unmeasured velocity. Indeed,

$$\dot{x}_j^* = \begin{cases} \sigma \alpha_j \rho_j + \sigma \alpha_j \delta_j e_\omega & \text{if } |s_j(q)| > \delta_K \\ 0 & \text{otherwise.} \end{cases} \quad (16)$$

where $\alpha = \text{diag}\{\alpha_1, \alpha_2, \alpha_3\}$ is a diagonal matrix, $\rho = [\rho_1 \ \rho_2 \ \rho_3]^\top$ and $\delta = [\delta_1 \ \delta_2 \ \delta_3]^\top$ are defined as

$$\begin{aligned} \alpha_j &= \frac{1}{2} \left[\frac{m_j \tau_d}{s_j} \right]^{-1/2} \\ \rho_j &= \frac{1}{s_j} \left[(k_d a + k_i) m_j \vartheta - k_i m_j e_q + \frac{\partial m_j}{\partial q} \tau_d \omega^* \right] - \frac{N_r c_j}{s_j^2} \tau_d \omega^* \\ \delta_j &= \frac{1}{s_j} \left[-m_j (k_p + k_d b) + \frac{\partial m_j}{\partial q} \tau_d \right] - \frac{N_r c_j}{s_j^2} \tau_d. \end{aligned}$$

Notice that α , ρ and δ are functions of known parameters and measurable variables. Since e_ω is not measurable, we introduce the following control law which is reminiscent of u defined in (14) except that we drop the term $\alpha \delta_j e_\omega$ in the definition of \dot{x}^* that is,

$$u = L(q)\sigma\alpha\rho + \omega^* K(q)x + Rx^* - k_{px} e_x, \quad (17)$$

where $e_x := x - x^*$. Note that, in view of (16), (17) is equivalent to

$$u = L(q)\dot{x}^* + \omega^* K(q)x + Rx^* - k_{px} e_x - \sigma L(q)\alpha \delta e_\omega \quad (18)$$

then, the closed-loop equation (2a) with (18) yields

$$L(q)\dot{e}_x = -[R + k_{px}] e_x - [K(q)x + \sigma L(q)\alpha \delta] e_\omega \quad (19)$$

which is also reminiscent of a perturbed linear system with stable drift; in this case, the input gain

$$g(t, x, \xi) := [K(q(t))x + \sigma L(q(t))\alpha(t, \xi)\delta(t, \xi)]$$

depends on known quantities and measured the states $\xi := [e_q \vartheta z]^\top$ and since L , K , m_j , $\frac{\partial m_j}{\partial q}$ and ω^* are uniformly bounded, there exists a non-decreasing function $\gamma_2 : \mathbb{R}_{\geq 0} \times \mathbb{R}_{\geq 0} \rightarrow \mathbb{R}_{\geq 0}$ such that

$$|g(t, x, \xi)| \leq \gamma_2 (|\xi|, |x|).$$

Therefore, it may be established that (19) is input-to-state stable with respect to the input e_ω , for an appropriate choice of the gain k_{px} dependent on γ_2 , hence on $|\xi|$ and $|x|$.

IV. MAIN RESULTS

The developments in the previous section indicate that the closed-loop system (9), (19) consists in the interconnection of two input-to-state stable systems for which the feedback gains may be adjusted to ensure global exponential stability. The proof of this claim constitutes our first contribution. Then, we show that in the case that all parameters are unknown, a certainty-equivalence adaptive controller ensures the convergence of the tracking errors to zero.

A. Robust control and system parameterization

Let $C(q) := \text{diag}\{c_j(q)\}$, $S(q) := \text{diag}\{s_j(q)\}$ then, $L(q) = \ell_0 I - \ell_1 C(q)$ and $K(q) = \ell_1 N_r S(q)$. With this notation, the control law (17) may be written as

$$u = \ell_0 \sigma \alpha \rho - \ell_1 C(q) \sigma \alpha \rho + \omega^* \ell_1 N_r S(q) x + R x^* - k_{px} e_x$$

which is linear in the physical parameters ℓ_0 , ℓ_1 and R therefore,

$$\begin{aligned} u &= \Psi(t, \xi)^\top \Theta - k_{px} e_x, \\ \Psi^\top &:= [\sigma \alpha \rho \quad \omega^* N_r S(q) x - C(q) \sigma \alpha \rho \quad x^*], \\ \Theta &:= \begin{bmatrix} \ell_0 & \ell_1 & R \end{bmatrix}^\top. \end{aligned} \quad (20)$$

We stress that Ψ is a function of t and the measured closed-loop states ξ ; indeed, one should read $q(t)$ and $x(t)$ in place of q and x while α , ρ and x^* are functions of t and ξ .

Proposition 1 *Consider the system (2) in closed loop with the controller (20), –see also (12) and (7). Let a , b , k_p , k_d , k_i and $\varepsilon > k_i$ be positive real numbers such that A in (9) is Hurwitz. Then, there exist a real number $k'_{px} > 0$ and a non-decreasing function $k''_{px} : \mathbb{R}_{\geq 0}^3 \rightarrow \mathbb{R}_{\geq 0}$ such that defining*

$$k_{px} = k'_{px} + k''_{px} (|\xi|, |x|, |x^*|), \quad (21)$$

the origin of the closed-loop system $\{[y_1, e_x] = [0, 0]\}$ is uniformly globally exponentially stable.

Proof: The rotor dynamical equation is equivalent to (6) for any τ_e^* , for the purpose of proof let $\tau_e^* = \tau_e^*(q, x^*)$ that is,

$$\tau_e^* := \frac{1}{2} \sum_{j=1}^3 K_j(q) |\hat{x}_j^*|^2$$

so (5) holds in view of the definition of x^* , this implies in turn that (6) is equivalent to (9). Now, a direct computation shows that,

$$|\tau_e - \tau_e^*| \leq \frac{k_M}{2} |e_x^\top (x + x^*)|. \quad (22)$$

On the other hand, substituting u from (20) in (2a) we obtain the closed-loop equation

$$L(q) \dot{e}_x = -[R + k_{px}] e_x - g(t, x, \xi) e_\omega. \quad (23)$$

Next, let $Q = Q_1 + Q_2$ where Q_1 is positive definite, $Q_2 = Q_2^\top \geq 0$ and there exists $q_m > 0$ such that $q_m |y_1|^2 \leq \frac{1}{2} y_1^\top Q_1 y_1$ and $y_1^\top Q_2 y_1 \geq \varepsilon_1 (b-1) e_\omega^2$. Let the Hurwitz assumption on A generate for such matrix Q , a positive definite matrix P and reals $p_M \geq p_m > 0$ such that V_1 in (10) satisfies

$$p_m |y_1|^2 \leq V_1(y_1) \leq p_M |y_1|^2$$

and the derivative of V_1 satisfies (11). Then, from (11) and (22) we obtain

$$\dot{V}_1(y_1) \leq -q_m |y_1|^2 + \frac{k_M \gamma_1}{2} |y_1| |e_x^\top (x + x^*)| - \varepsilon_1 (b-1) e_\omega^2.$$

Next, consider the function $V_2 : \mathbb{R}_{\geq 0} \times \mathbb{R}^3 \rightarrow \mathbb{R}_{\geq 0}$ defined by

$$V_2(t, e_x) = \frac{1}{2} e_x^\top L(q(t)) e_x$$

which is positive definite and radially unbounded since $|L(q)|$ is uniformly bounded and positive definite actually, in view of (3),

$$\frac{\ell_m}{2} |e_x|^2 \leq V_2(t, e_x) \leq \ell_M |e_x|^2.$$

The total derivative of V_2 along the trajectories of (23) satisfies

$$\begin{aligned} \dot{V}_2(t, e_x) &\leq -e_x^\top \left[(k_{px} - \gamma_2^2 \lambda_1 - \frac{k_M}{2} \omega^*) I \right. \\ &\quad \left. - \frac{\lambda_3}{2} e_x^\top K(q) e_x K(q) \right] e_x + \left(\frac{1}{\lambda_1} + \frac{1}{\lambda_3} \right) e_\omega^2 \end{aligned}$$

for which we used $\widehat{L(q)} = K(q)[e_\omega + \omega^*]$, $e_x^\top K(q) e_x e_\omega \leq (1/\lambda_3) e_\omega^2 + \lambda_3 [e_x^\top K(q) e_x]^2$ and $e_x^\top g e_\omega \leq \lambda_1 \gamma_2^2 |e_x|^2 + (1/\lambda_1) e_\omega^2$. Thus, the total derivative of the Lyapunov function $V_3 := V_1 + V_2$ along the closed-loop trajectories satisfies

$$\begin{aligned} \dot{V}_3 &\leq -\left[\varepsilon_1 (b-1) - \left(\frac{1}{\lambda_1} + \frac{1}{\lambda_3} \right) \right] e_\omega^2 - \left[q_m - \frac{\lambda_2 k_M \gamma_1}{2} \right] |y_1|^2 \\ &\quad - \left[k_{px} - \lambda_1 \gamma_2^2 - \frac{k_M}{2} (\omega^* + \lambda_3 k_M |e_x|^2 \right. \\ &\quad \left. + \frac{\gamma_1}{\lambda_2} |x + x^*|^2) \right] |e_x|^2 \end{aligned} \quad (24)$$

We see that for any given q_m , ε_1 and b there exist positive numbers λ_1 , λ_2 , λ_3 and c such that defining $k'_{px} > 0$ and²

$$\begin{aligned} k''_{px} &:= \frac{k_M}{2} (\omega^* + \lambda_3 k_M |e_x|^2 + \frac{\gamma_1}{\lambda_2} |x + x^*|^2) \\ &\quad + \lambda_1 \gamma_2 (|\xi|, |x|)^2 \end{aligned} \quad (25)$$

we have

$$\dot{V}_3(t, y_1, e_x) \leq -c \left[|y_1|^2 + |e_x|^2 \right]. \quad (26)$$

The result follows. ■

²Notice that $|x^*|^2 = \mathcal{O}(\tau_d)$ hence, it is bounded by a non-decreasing function of order $\mathcal{O}(|\xi|)$.

B. Control under full parametric uncertainty

In the previous section we assumed that the rotor inertia J and the load-torque are unknown and we established uniform global exponential stability of the origin. The latter includes $\{[y_1 \ e_x] \in \mathbb{R}^7 : e_\omega = 0, e_x = 0\}$. This implies that currents follow exponentially *any* reference x^* as defined in (12) and the rotor angular velocity is stabilized at the desired reference ω^* . Note that this holds regardless of the value of σ . Now we relax the assumption that L and R are known that is, we assume that the physical parameters R, ℓ_0, ℓ_1 and hence $\sigma(\ell_1)$, are also unknown.

Let us redefine the reference for the stator currents by using the *constant*

$$\hat{\sigma} = \sqrt{\frac{2}{N_r \hat{\ell}_1(0)}}$$

where $\hat{\ell}_1(0)$ denotes the best estimate available of ℓ_1 . That is, we use the reference $\hat{x}^* := x^*(\hat{\sigma})$ where $x^*(\cdot)$ is defined in (12) and to avoid an over-parameterization, we choose to estimate online only $\Theta := [\ell_0, \ell_1, R]$.

Proposition 2 Consider the system (2) in closed loop with the certainty-equivalence controller

$$u = \Psi(t, \xi, \hat{e}_x)^\top \hat{\Theta} - k_{px} \hat{e}_x, \quad \hat{e}_x := x - \hat{x}^* \quad (27)$$

and the adaptation law

$$\dot{\hat{\Theta}} = -k_\theta \Psi(t, \xi, \hat{e}_x) \hat{e}_x, \quad k_\theta > 0 \quad (28)$$

under the conditions of Proposition 1 with \hat{x}^* in place of x^* in (21). Define $\tilde{\Theta} := \hat{\Theta} - \Theta$. Then, the origin of the closed-loop system, $\{[y_1, \hat{e}_x, \tilde{\Theta}] = [0, 0, 0]\}$ is uniformly globally stable and the tracking errors y_1 and \hat{e}_x satisfy

$$\lim_{t \rightarrow \infty} |y_1(t)| = 0 \quad \lim_{t \rightarrow \infty} |\hat{e}_x(t)| = 0.$$

Proof: The closed-loop system is given by equations (9), (23) with \hat{e}_x in place of e_x . Consider the Lyapunov function $V : \mathbb{R}_{\geq 0} \times \mathbb{R}^4 \times \mathbb{R}^3 \times \mathbb{R}^3 \rightarrow \mathbb{R}_{\geq 0}$ defined by

$$V(t, y_1, \hat{e}_x, \tilde{\Theta}) := V_3(t, y_1, \hat{e}_x) + \frac{1}{2k_\theta} |\tilde{\Theta}|^2$$

which is positive definite and radially unbounded under the conditions of Proposition 1. Following the steps of proof of Proposition 1, using

$$u = \Psi(t, \xi, \hat{e}_x)^\top \hat{\Theta} - k_{px} \hat{e}_x + \Psi(t, \xi, \hat{e}_x)^\top \tilde{\Theta},$$

and replacing x^* in (24) and (25) with \hat{x}^* , yields

$$\dot{V}(t, y_1, \hat{e}_x, \tilde{\Theta}) \leq -c \left[|y_1|^2 + |\hat{e}_x|^2 \right] \leq 0 \quad (29)$$

in place of (26). Uniform global stability of the origin follows by integrating $\dot{V} \leq 0$ along the closed-loop trajectories. In addition, by integrating the first inequality in (29) we see that $y_1 \in \mathcal{L}_2, \hat{e}_x \in \mathcal{L}_2$. Furthermore, uniform global stability implies uniform global boundedness hence $y_1 \in \mathcal{L}_\infty, \hat{e}_x \in \mathcal{L}_\infty$; in view of (3), a simple inspection at the closed-loop equations shows that $\dot{y}_1 \in \mathcal{L}_\infty$ and $\dot{\hat{e}}_x \in \mathcal{L}_\infty$. The result follows, invoking Barbalát's lemma. ■

V. SIMULATION RESULTS

With aim at evaluating the controller of Proposition 2 we have performed some numerical simulations in SIMULINK™ of MATLAB™, using the parameters presented in [10] $R = 0.3 [\Omega], \ell_0 = 24 [\text{mH}], \ell_1 = 19 [\text{mH}]$ and $N_r = 25$. The control gains are $a = 1500, b = 3200, k_p = 1050, k_i = 5 \times 10^{-4}, k_d = 1000, k'_{px} = 150, k''_{px} = 0$ and $k_{\theta 1} = 15$. We stress that for implementation purpose, we use a constant value of k_{px} even though the sufficient condition for *global* stability is that this gain depends on the measured states. The reason to fix $k''_{px} = 0$ is to avoid high values in the input voltages which bring the converters into saturation.

The experiment consist in imposing a smoothed step speed reference defined by

$$\omega^*(t) = \omega_0^* + \frac{(\omega_f^* - \omega_0^*)}{2} (1 + \tanh(t - T)) \quad (30)$$

with the final desired velocity set on $190 [\text{rad/s}]$. The load torque is piecewise constant, for the first half of the simulation time $\tau_L = 0.1$ then, it is increased by 50% at $t = 3.5\text{s}$.

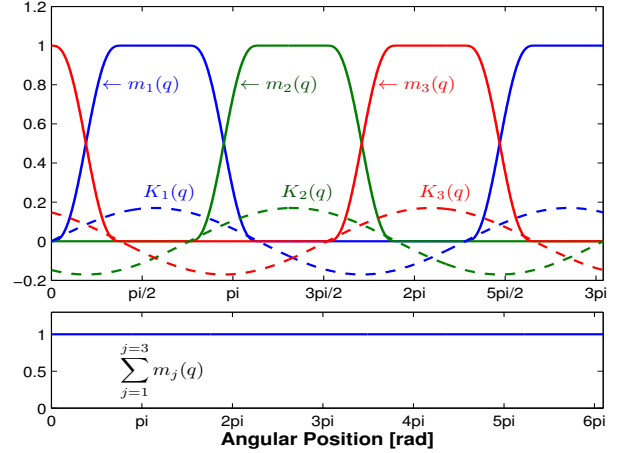


Fig. 1. Graph of the commutation functions $q \mapsto m_j$

The commutation functions $m_j : [0, 2\pi) \rightarrow \mathbb{R}_{\geq 0}$ are illustrated in Fig. 1 and are defined as follows. Let

$$f(x) = 10 \frac{x^3}{\theta_m^3} - 15 \frac{x^4}{\theta_m^4} + 6 \frac{x^5}{\theta_m^5}$$

then, for $j = 1, 2, 3$ and $q_1 = \text{mod}(q_0, \pi/4), q_2 = \text{mod}(q_0 - \pi/12, \pi/4), q_3 = \text{mod}(q_0 + \pi/12, \pi/4)$,

$$m_j^+(\theta) = \begin{cases} f_j(q_{aj}) & 0 < q_{aj} \leq \frac{\pi}{3N_r} \\ 1 & \frac{\pi}{3N_r} < q_{aj} \leq \frac{2\pi}{3N_r} \\ 1 - f_j(q_{aj}) & \frac{2\pi}{3N_r} < q_{aj} \leq \frac{\pi}{N_r} \\ 0 & \text{otherwise} \end{cases} \quad (31)$$

$$m_j^-(\theta) = \begin{cases} f_j(q_{aj}) & \frac{\pi}{N_r} < q_{aj} \leq \frac{4\pi}{3N_r} \\ 1 & \frac{4\pi}{3N_r} < q_{aj} \leq \frac{5\pi}{3N_r} \\ 1 - f_j(q_{aj}) & \frac{5\pi}{3N_r} < q_{aj} \leq \frac{2\pi}{N_r} \\ 0 & \text{otherwise} \end{cases} \quad (32)$$

The operator mod resets q i.e., $q_j = \text{mod}(\beta_1, \beta_2)$ is such that $q_j(0) = \beta_1$ and q_j is reset to the latter when $q_j(t) = \beta_2$.

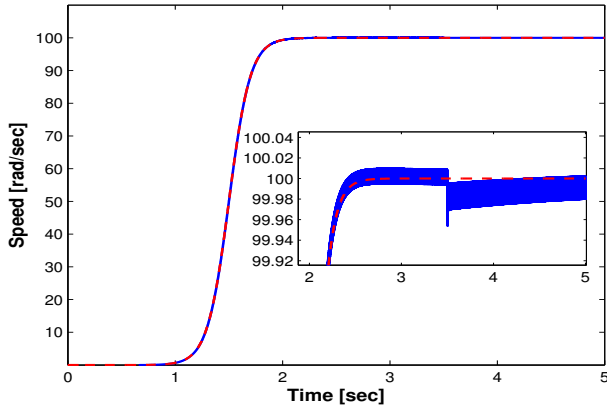


Fig. 2. Velocity Tracking

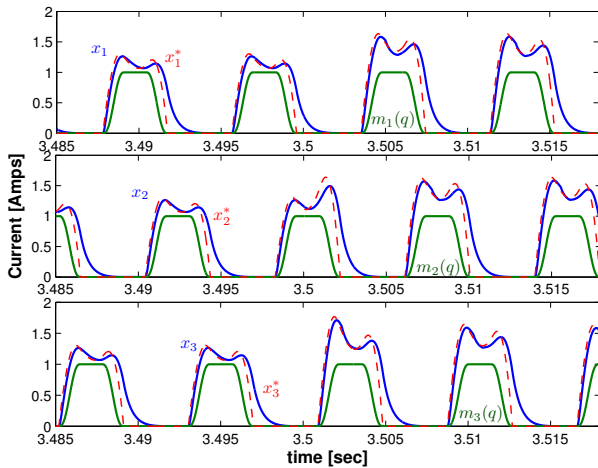


Fig. 3. Electric current around $t = 3.5s$ when the load-torque increases. The actual currents x_j as well as the reference trajectories \hat{x}^* and the commutation signals $m_j(q(t))$ are showed to be in synchrony

In Fig. 2 is depicted the system's response; note the perfect tracking of the angular velocity and the good performance. In Fig. 3 we show a zoomed window on the three stator currents and their references. The commutations due to the sharing torque approach may be clearly appreciated, as well as the effect of the load torque increase at $t = 3.5s$, the motor requires more electrical current in order to remain at the required velocity set-point. The corresponding input voltages are depicted in Fig. 4.

VI. CONCLUDING REMARKS

We presented an adaptive controller for the switched-reluctance motor, considering both the stator and rotor dynamics. Our control approach exploits the physical properties of the machine, relying on torque-sharing approach. Ongoing research focuses on the sensorless control problem that is, avoiding the use of position measurements.

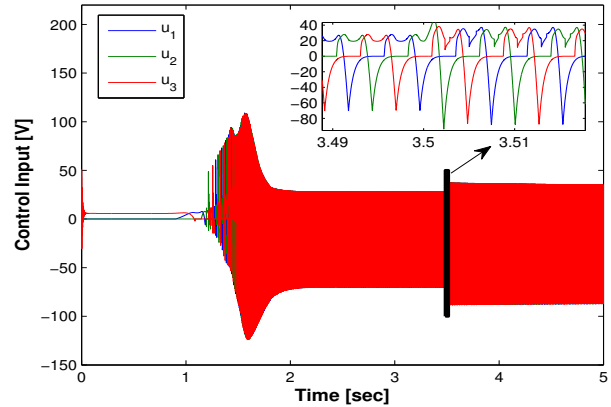


Fig. 4. Voltage control inputs u_j

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