Transfer function



Some typical transfer functions:

1. First-order with delay process, $G(s)=k e^{-\theta s}/(\tau s+1)$

=

- Example: Heated tank with delay in heater, y=T, u=Q, V dT/dt = $q_{in}(T_{in}-T(t)) + Q(t-\theta)/(\rho c_P)$ Get $\tau = V/q_{in}$ and k=1/($\rho c_P V$)
- 2. Integrating process, G(s)=k'/s

Example: y = level (V) and u=qin, dV/dt = q_{in} - q_{out} Get k'=1

3. PID-controller , C(s) = $K_c(1 + 1/(\tau_l s) + \tau_D s)$ («ideal PID»)

$$K_c \frac{\tau_I \tau_D s^2 + \tau_I s + 1}{\tau_I s}$$

1

First-order system

y(s) = G(s) u(s)

Two standard forms of first-order system:

1. G(s) = b/(s-a) where a = pole («state space» form) Follow from general case with A=a, B=b, C=1, D=0. Note: k'=b=initial slopeTime response to step M in u(t): $y(t) = \frac{bM}{-a} (1 - e^{at})$ • Stable for a<0 since exp(at) =0 as t->∞ • Unstable for a>0 since exp(at) -> ∞ as t->∞

2. $G(s) = k/(\tau s+1)$ (time constant form for stable system) $\tau = -1/a$ k=g(0) = -b/a = steady-state gain

Time response to step M in u(t): $y(t) = kM (1 - exp(-t/\tau))$

General procedure, matrix state-space form

1. General* Nonlinear dynamic model:

dx/dt = f(x,u,d), y = g(x,u,d) (x is vector of states, y is vector of "outputs")

- 2. Steady state model. $dx^*/dt=0 \rightarrow f(x^*,u^*,d^*)=0$
 - Find steady state *. Typically, use to find missing data
- 3. Introduce deviation variables and linearize

$$- dx/dt = \Delta f = A \Delta x(t) + B \Delta u(t) + B_d \Delta d(t). \quad A = \left(\frac{\partial f}{\partial x}\right)_* \text{ etc.}$$

$$- \qquad \Delta y(t) = \Delta g = C \Delta x(t) + D \Delta u(t) + D_d \Delta d(t)$$

- 4. Laplace** of both sides of linear model* (t -> s)
 - sx(s) = A x(s) + B u(s) + B_d d(s)

$$-$$
 y(s) = C x(s) + D u(s) + D_d d(s)

- 5. Algebra (eliminate x(s), see next page)
 - y(s) = G(s) u(s) + G_d d(s)

- Transfer matrix,
$$G(s) = C (sI-A)^{-1}B + D$$
,

$$G_{d}(s) = C (sI-A)^{-1}B_{d} + D_{d}$$

- 6. Block diagram
- 7. Controller design

*State-space form (differential equations) is not completely general:

1) Cannot handle time delay.

2) Let g(s)= n(s)/d(s). Must assume: order $d(s) \ge order n(s)$ (so cannot handle ideal PID) **We will only use Laplace for linear systems!

General* Transfer Matrix

General system with n differential equations in n state variables x(t) (where x, u, y are vectors and A, B, C, D are matrices):

$$\frac{dx(t)}{dt} = A x(t) + B u(t)$$
$$y(t) = C x(t) + D u(t)$$

Laplace transform with zero intitial condition, x(0) = 0, u(0) = 0 (deviation variables):

$$sI x(s) = A x(s) + B u(s)$$

(sI - A) x(s) = B u(s)
x(s) = (sI - A)^{-1} B u(s)

Get y(s) = G(s)u(s) where transfer matrix is:

$$G(s) = C (sI - A)^{-1} B + D$$

Here

$$(sI - A)^{-1} = \frac{adj(sI - A)}{det(sI - A)}$$

where det(SI - A) =

$$d(s) = a_n s^n + a_{n-1} s^{n-1} + \dots + a_1 s + a_0$$

is a n'th order polynomial in n,

The n roots (generally complex) of the polynomial d(s), d(s) = det(sI-A)=0 are the same as the eigenvalues of the state matrix A, and are known as the «poles» of the system. A system with n states (so A is a nxn matrix) has n roots = eigenvalues = poles

Plan for next two weeks

1. First-order systems (SiS5)

3. Second-order systems

- Can have oscillations (complex poles)
- 4. Closed-loop transfer function (with control)

5. Poles and zeros

- Including inverse response (RHP-zeros)
- 6. Slow oscillations for PI-control of integrating process

7. Approximating transfer functions

- Time delay
- Half rule

8. Derivation of SIMC PID rules (SiS6)

Initial and final values for step response

- Transfer function g(s)
 - <mark>y(s) = g(s) u(s)</mark>
- Deviation variables for y(t) and u(t)
- Consider response y(t) to **step** of magnitude M in input.
 - $u(t) = 0 \text{ for t>0, } u(t) = M \text{ for t≥0} \Rightarrow u(s)=M/s$
- From g(s) we get directly final and intital part of time response:

Steady-state gain:
$$\frac{y(\infty)}{M} = g(0)$$

Initial gain: $\frac{y(0^+)}{M} = g(\infty)$
Initial slope: $\frac{y'(0^+)}{M} = \lim_{s \to \infty} sg(s)$
Proof: Note that $y(s) = g(s)\frac{M}{s}$
Final value theorem: $\lim_{t \to \infty} y(t) = \lim_{s \to 0} sy(s) = \lim_{s \to 0} sg(s)\frac{M}{s} = g(0)M$
Initial value theorem: $\lim_{t \to 0} y(t) = \lim_{s \to \infty} sy(s) = g(\infty)M$
Initial value theorem: $\lim_{t \to 0} y'(t) = \lim_{s \to \infty} s(sy(s)) = \lim_{s \to \infty} sg(s)M$
Initial value theorem: $\lim_{t \to 0} y^{(n)}(t) = \lim_{s \to \infty} s^n(sy(s)) = \lim_{s \to \infty} s^ng(s)M$



First-order system responses

Time response: $y(t) = kM (1 - exp(-t/\tau))$

Integrating system, g(s)=k'/s.

Special case of first-order system with $i \ge \infty$ and $k = \infty$ but slope k' = k/i is finite

2nd order system. Special case: Two first-order in series $G(s) = \frac{K}{(\tau_1 s+1)(\tau_2 s+1)}$

Example: Temperature in two tanks in series, $\tau_1 = V_1/q$, $\tau_2 = V_2/q$

(1) y(s) = G(s) u(s) with u(s)=M/s (step). Partial fraction expansion of (1) (2) $y(s) = kM \left(\frac{C_0}{s} + \frac{C_1}{s+1/\tau_1} + \frac{C_2}{s+1/\tau_2}\right)$ where $C_0=1, C_1=-\tau_1/(\tau_1-\tau_2), C_2=\tau_2/(\tau_1-\tau_2)$ (Find C_0, C_1 and C_2 from (1)=(2)).

Inverse Laplace of (2) y(t) =kM ($C_0+C_1e^{-t/\tau 1}+C_2e^{-t/\tau 2}$)

Conclusion: Step response (M = change in input):

$$y(t) = KM\left(1 - \frac{\tau_1 e^{-t/\tau_1} - \tau_2 e^{-t/\tau_2}}{\tau_1 - \tau_2}\right) \quad (5-47)$$

Example G1(s) = 2/(2s+1). m=n_p=1. So first derivative y'(t) (initial slope) is <u>non</u>-zero

Example $G7(s) = 2/(2s+1)^7$. m=n_p=7. So six first derivatives of y(t) are zero -> Very flat initial response. Almkost like time delay.

General 2nd order system

$$G(s) = \frac{K}{\tau^2 s^2 + 2\zeta \tau s + 1} = \frac{K}{\tau^2 (s - \lambda_1)(s - \lambda_2)}$$

Roots (poles, eigenvalues): $\lambda_{1,2} = \frac{-\zeta \pm \sqrt{\zeta^2 - 1}}{\tau}$

- $\zeta > 1$ Overdamped (two real poles)
- $\zeta = 1$ Critically damped (two real identical poles)
- $|\zeta| < 1$ Underdamped (complex poles; oscillations)

 $\zeta < 0$ Unstable

1. Two real poles (overdamped), ³ >1: Two first-order in series

$$G(s) = \frac{K}{(\tau_1 s + 1)(\tau_2 s + 1)} = \frac{K}{\tau_1 \tau_2 s^2 + (\tau_1 + \tau_2) s + 1} \qquad \zeta = \frac{\tau_1 + \tau_2}{2\sqrt{\tau_1 \tau_2}} \ge 1$$
$$\tau = \sqrt{\tau_1 \tau_2}$$

Two - real - poles: $\lambda_1 = -1/\tau_1, \lambda_2 = -1/\tau_2 \qquad y(t) = KM\left(1 - \frac{\tau_1 e^{-t/\tau_1} - \tau_2 e^{-t/\tau_2}}{\tau_1 - \tau_2}\right) \quad (5-47)$

2. Complex poles, $|^3| < 1$

$$G(s) = \frac{K}{\tau^2 s^2 + 2\zeta \tau s + 1} = \frac{K}{\tau^2 (s - \lambda_1)(s - \lambda_2)}$$

Poles:
$$\lambda_{1,2} = \frac{-\zeta \pm \sqrt{\zeta^2 - 1}}{\tau} = \sigma \pm i\omega$$

where $\sigma = \frac{-\zeta}{\tau}, i = \sqrt{-1}, \omega = \frac{\sqrt{1 - \zeta^2}}{\tau}$

y(s) = G(s) u(s) with u(s)=M/s (step).

Inverse Laplace

$$y(s) = kM \left(\frac{C_0}{s} + \frac{C_1}{s - \lambda_1} + \frac{C_2}{s - \lambda_2}\right) \rightarrow y(t)/kM = C_0 + C_1 e^{\lambda t} + C_2 e^{\lambda 2t}$$

Complex poles give oscillations! Use Euler's formula for complex parts: $e^{i\omega t} = \cos \omega t + i \sin \omega t$ The complex parts in y(t) cancel! Finally get for y(t)/kM:

21.
$$1 - \frac{1}{\sqrt{1 - \zeta^2}} e^{-\zeta t/\tau} \sin\left[\sqrt{1 - \zeta^2} t/\tau + \psi\right]$$

 $\psi = \tan^{-1} \frac{\sqrt{1 - \zeta^2}}{\zeta}, \quad (0 \le |\zeta| < 1)$

Note: $\sqrt{1-\zeta^2} = \omega \tau$ where ω is complex part of pole

Simpler form:

$$y(t) = kM(1 - \frac{1}{\omega\tau}e^{\sigma t}\sin(\omega t + \psi))$$

$$\psi = \tan^{-1}(\frac{\omega}{\sigma})$$

$$\omega/\sigma = \text{Complex-part/Real-part}$$

Table 3.1 Laplace Transforms for Various Time-DomainFunctions^a (continued)

f(t)	F(s)	
20. $1 + \frac{1}{\tau_2 - \tau_1} (\tau_1 e^{-t/\tau_1} - \tau_2 e^{-t/\tau_2})$	$\frac{1}{s(\tau_1 s + 1)(\tau_2 s + 1)}$	
$21. \ 1 - \frac{1}{\sqrt{1 - \zeta^2}} e^{-\zeta t/\tau} \sin\left[\sqrt{1 - \zeta^2} t/\tau + \psi\right]$ $y_t = \tan^{-1} \sqrt{1 - \zeta^2} (0 \le \zeta < 1)$	$\frac{1}{s(\tau^2 s^2 + 2\zeta\tau s + 1)}$	Alternative forms of step
$\psi = \tan \zeta , (\sigma = \varsigma = \gamma$ $22. 1 - e^{-\zeta t/\tau} [\cos(\sqrt{1 - \zeta^2} t/\tau) + \frac{\zeta}{\sqrt{1 - \zeta^2}} \sin(\sqrt{1 - \zeta^2} t/\tau)]$	$\frac{1}{s(\tau^2 s^2 + 2\zeta \tau s + 1)}$	Simpler form: $y(t) = kM(1 - \frac{1}{\omega\tau}e^{\sigma t}\sin(\omega t + \psi))$ $\psi = \tan^{-1}(\frac{\omega}{\sigma})$
$\sqrt{1-\zeta^2}$ $(0 \le \zeta < 1)$		
23. $1 + \frac{\tau_3 - \tau_1}{\tau_1 - \tau_2} e^{-t/\tau_1} + \frac{\tau_3 - \tau_2}{\tau_2 - \tau_1} e^{-t/\tau_2}$	$\frac{\tau_{3}s+1}{s(\tau_{1}s+1)(\tau_{2}s+1)}$	
$(\tau_1 \neq \tau_2)$		
24. $\frac{df}{dt}$	sF(s) - f(0)	
25. $\frac{d^n f}{dt^n}$	$s^{n}F(s) - s^{n-1}f(0) - s^{n-2}f^{(1)}(0) - \cdots$ - $sf^{(n-2)}(0) - f^{(n-1)}(0)$	
26. $f(t-t_0)S(t-t_0)$	$e^{-t_0s}F(s)$	

^{*a*}Note that f(t) and F(s) are defined for $t \ge 0$ only.

Figure 5.8. Step response of underdamped second-order processes.

s=tf('s')
zeta=0.5, tau=1
g = 1/[(tau*s)^2 + 2*tau*zeta*s + 1]
step(g,20)

S=tf('s') <mark>zeta=0,</mark> tau=1 g = 1/[(tau*s)^2 + 2*tau*zeta*s + 1]

 $G(s) = \frac{K}{\tau^2 s^2 + 2\zeta \tau s + 1}$ |ζ|<1

Poles: $\lambda_{1,2} = \sigma \pm i\omega$ $\sigma = \frac{-\zeta}{\tau}, \quad \omega = \frac{\sqrt{1-\zeta^2}}{\tau}$ $t_p = \pi/\omega$ OS=exp(πσ/ω) $DR = (OS)^2 = \exp\left(-2\pi\zeta/\sqrt{1-\zeta^2}\right)$ $DR = OS^2$ (5-52)

 $P=2t_p=2\pi/\omega$

Period:
$$P = \frac{2\pi\tau}{\sqrt{1-\zeta^2}}$$
(5-53)

c/a= Decay ratio:

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Example 1

$$T(s) = \frac{4}{(4s+1)(s+1)+4} = \frac{0.8}{0.8s^2+s+1} = \frac{k}{\tau^2 s^2 + 2\tau \zeta s + 1}$$

Get

$$\tau = \sqrt{0.8} = 0.8944s, \quad \zeta = \frac{1}{2 \cdot 0.8944} = 0.559$$

 $\sigma = \frac{-\zeta}{\tau} = -0.625, \quad \omega = \frac{\sqrt{1-\zeta^2}}{\tau} = 0.927$

Time to first peak: $t_p = \pi/\omega = 3.14/0.927 = 3.39$ s Overshoot: OS = $\exp(\pi\sigma/\omega) = exp(-2.12) = 0.120$ Period: $P = 2\pi/\omega = 2t_p = 6.78$ s Time to second peak: $3t_p = 10.17$ s

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COMPLEX POLES IN PRACTISE

Underdamped (Oscillating) second-order systems (|³|<1)

$$G(s) = \frac{K}{\tau^2 s^2 + 2\zeta \tau s + 1}$$

Corresponds to complex poles, $\ \lambda_{1,2}=\sigma\pm i\omega$

Process systems: Oscillations are usually caused by (too) aggressive control U-tube is an exception (see example next slide)

Example 1: P-control (controller gain K_c) of second-order process, g(s) = k/(i_1 s+1)(i_2 s+1)

• Oscillates (³ < 1) if K_ck is <u>large</u>

But there also cases where we need «aggressive» control to avoid oscillations:

Example 3: PI-control of integrating process, g(s)=k'/s

- Need control to stabilize
- Oscillates (³<1) if K_ck' is <u>small (!)</u>

29/9-23 Example naturally oscillation process: V-tube PIDNIN?] sheady-state leve TE THEMT ~ 50 cm (L=1m) Newtons 2nd law (F=ma) FA - FP2 - Fg - Ffriction = M at Hore FPI=P.A FP2=P2A Fg = Og. 2h.A Ft = KFV (laminar flow). Also note that V= al $(\underline{P}_{1}-\underline{P}_{2})A - \underline{P}_{3} 2\underline{A}_{h}^{(t)} - \underline{k} \frac{dh^{(t)}}{dx} = \underline{m} \frac{d^{2}h}{dt}$ Ap(t) Note: The50 get ms2h02 kesh02 2001h(1)=Adp(5) Laplace h(s) = (A Apls) Oscillates for B2-44(20 ⇒ k2-80gmACO =) [kg < V80gmA] it friction is small.

30/9-23 On standard form 50 C= V200A = VII = 5 = Ky200 A = Kr = KA (decillates for SLI = ke < Vspgn Ales found above) Friddin forms The App A where $\Delta P_{t} = \frac{1}{2} \frac{1$ THE STALY Alamarial exarche $L = 1 m_0 = 0.01 m_1$ $L = 1 m_0 = 0.01 m_2$ $A = \frac{2}{10} p^2 = 0.78 \cdot 10^{-4} m_2^2$ M= 10-3 10, (mater) $Q = 10^{3} \text{ kg/m}(-1)$ Get $\tau = \sqrt{\frac{1}{2\pi}} = \sqrt{\frac{1}{2\pi}} = 0.225$ (fast)) P=2 $\pi\tau$ = 1.40s 5 = kr PAV85L = 8TML = TV 8L M = 3.14 10 103 81 04 = 0.036 [ascillets!!! To get vid of the oscillations we need to have a contraction (smaller A) or a viscous fluid (larger M) For example, reducing 0 by a factor 10 (from 1cm to 1mm) reducing A by a factor 100 and we get 5 = 3,6 (no ascillations

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Example 1. Setpoint response for P-control of 2nd order process

g(s) = k/[(4s+1)(s+1)]P-controller: $c(s) = K_c$

Task: Derive closed-loop transfer function T(s) for setpoint change (can set d=0).

Note: same example as before

 $g(s) = k/[(4s+1)(s+1)], g_m=1$ P-controller: $c(s) = K_{c_1} Data: K_c k = 1$

Example 1

0

$$\begin{split} T(s) &= \frac{4}{(4s+1)(s+1)+4} = \frac{0.8}{0.8s^2 + s + 1} = \frac{0.8}{\tau^2 s^2 + 2\tau \zeta s + 1}\\ \text{Get} \\ \tau &= \sqrt{0.8} = 0.8944s, \quad \zeta = \frac{1}{2 \cdot 0.8944} = 0.559\\ \sigma &= \frac{-\zeta}{\tau} = -0.625, \ \omega = \frac{\sqrt{1-\zeta^2}}{\tau} = 0.927 \end{split}$$

Time (seconds)

Time to first peak: $t_p = \pi/\omega = 3.14/0.927 = 3.39$ s Overshoot: OS = $\exp(\pi\sigma/\omega) = exp(-2.12) = 0.120$ Period: $P = 2\pi/\omega = 2t_p = 6.78 \text{ s}$ Time to second peak: $3t_p = 10.17$ s

Can also find ω from poles $0.8 s^2 + s + 1 = 0$ Solution: $s = -0.625 \pm 0.927i$

Closed-loop transfer function

(1) Process: $y = g(s) u + g_d(s) d$ (2) Controller: $u = c(s) (y_s - y_m)$ (3) Measurement: $y_m = g_m(s) y + n$

Closed-loop response: Want to find effect of y_s , d an n on output y. Task: Eliminate u and y_m to find $y = T(s) y_s + T_d(s) d + T_n(s) n$

Closed-loop transfer functions

- Closed-loop output response: $y = T y_s + T_d d + T_n n$
- Introduce «loop»= L = g c g_m
 - T(s) = gc/(1+L) $T_d(s) = g_d/(1+L)$

 $T_n = -T$

- General rule for negative feedback:
 - Transfer function = «direct(s)» / (1 + «loop(s)»)
- Example: What is response from y_s, d and n to u?

$$- u = T_{us} y_s + T_{ud} d + T_{un} n$$

- Note that $T_{ud} = -T$ for input disturbance ($g_d=g$) and $g_m=1$
 - which is interesting since with SIMC-rule we choose desired T, so we also «have control» over input change for disturbances

Sensitivity function S

S gives effect of feedback

- T_{cl} = Direct*S where S=1/(1+loop)
 - No control: S=1 (S=I for multivariable case)
 - Want |S| small to have small control error |e|:
 e = S (y_s g_m g_d d)
 - Perfect control (infinite c): S=0 (Achievable at steady state with I-action)

Steady-state offset with P-control

(k=process gain, K_c= controller gain)

Sensitivity fuction: S(s) = 1/(1+L) where $L(s)=loop = g c g_m$. S is transfer function from y_s to control error e (since «Direct»=1): $e = S(s) y_s$ Steady-state offset to step change in setpoint: $e = S(0) y_s$ where $S(0)= 1/(1+loop(0))= 1/(1+K_ck)$ since g(0)=k, $c(0)=K_c$, $g_m(0)=1$

Example P-control. k=1, K_c = 4. Relative Steady-state offset e/y_s is S(0)=1/(1+K_ck) = 1/5= 0.20 (20%)

Note with I-action (PI-control): Get $c(0)=\infty$, $loop(0)=\infty$, so S(0)=0 and get no steady-state offet.

Example 2. PI-control of 1st order process

Example.

 $g(s) = g_{d}(s) = 2/(3s+1)$ SIMC PI-controller, $\tau_{c}=0.5$ (so we are speeding up the response relative to $\tau=3$): $c(s) = K_{c}(1 + \frac{1}{\tau_{I}s}), \quad K_{c}=3, \quad \tau_{I}=2$ $-> \quad c(s) = 3\left(1 + \frac{1}{2s}\right) = 3\frac{2s+1}{2s}$

5))

Input Disturbance response (g_d=g)

Example 3. PI-control of integrating process (level)

Task: Derive condition to avoid «slow» ocillations that may occur when K_c is too small

*Yes, this may seem a bit strange, but for PI-control of integrating process you may get oscillatons when Kc is too small! In addition, you may of course get the more common «fast» oscillations if Kc is too large because of «overreaction» with time delay. Integrating process with PI-control:

$$G(s) = \frac{k'}{s}$$
$$C(s) = K_c (1 + \frac{1}{\tau_I s})$$

0

General rule to avoid slow oscillations ($\zeta \ge 1$) : $k' K_C \tau_I \geq 4$

Need large controller gain and/or large integral time (!)

Proof:

$$G(s) = k \frac{e^{-\theta s}}{\tau_1 s + 1} \approx \frac{k'}{s} \text{ where } k' = \frac{k}{\tau_1}; \ C(s) = K_c \left(1 + \frac{1}{\tau_I s}\right)$$

Closed-loop poles:
$$1 + GC = 0 \Rightarrow 1 + \frac{k'}{s} K_c \left(1 + \frac{1}{\tau_I s}\right) = 0 \Rightarrow \tau_I s^2 + k' K_c \tau_I s + k' K_c = 0$$

To avoid oscillations we must not have complex poles:
$$B^2 - 4AC \ge 0 \Rightarrow k'^2 K_c^2 \tau_I^2 - 4k' K_c \tau_I \ge 0 \Rightarrow k' K_c \tau_I \ge 4 \Rightarrow \tau_I \ge \frac{4}{k' K_c}$$

Closed-loop responses

Closed-loop response to disturbance d at input and setpoint change $y = \frac{q}{1+q_c}d + \frac{q_c}{1+q_c}y_s$ PI-control of integrator: $g(s) = \frac{1}{s}; \quad c(s) = K_c \frac{\tau_{I}s+1}{\tau_{I}s}$ Get $y = \frac{\tau_{Is}}{\tau_{I}s^2 + K_c \tau_{I}s + K_c}d + \frac{K_c(\tau_{I}s+1)}{\tau_{I}s^2 + K_c \tau_{I}s + K_c}y_s$ With $\tau_I = 1, K_c = 0.25:$ $y = \frac{s}{s^2 + 0.25s + 0.25}d + \frac{0.25(s+1)}{s^2 + 0.25s + 0.25}y_s = \frac{4s}{4s^2 + s + 1}d + \frac{(s+1)}{4s^2 + s + 1}y_s$ Note: This is not a good tuning, Will get slow oscillations since $k'K_c\tau_l = 1*0.25*1=0.25 < 4$

Notes:

- Steady-state gain h(0) for disturbance transfer function h(s) is zero (because controller has integral action)
- Steady-state gain T(0) for setpoint transfer function T(s) is 1 (because controller has integral action)
- Denominator is on form $\tau^2 s^2 + 2\tau \zeta s + 1$ with $\tau = 2$ and $\zeta = 0.25 < 1$, so there will be oscillations with period $P \approx 2\pi\tau$
- Initial response (t → 0) to disturbance is the same as with no control
 (h(s) = g/(1+gc) → g(s) when s → ∞ since g(s)c(s) → 0 (which is the case
 for all real systems))

Simulink, tunepid4

Function Block Parameters: Transfer Fcn1	×
Transfer Fcn	
The numerator coefficient can be a vector or matrix expression. The denominator coefficient must be a vector. The output width equals the number of rows in the numerator coefficient. You should specify the coefficients in descending order of powers of s.	
Parameters	
Numerator coefficients:	
Kc * [taui*taud taui+taud 1]	
Denominator coefficients:	
[0.01*taui*taud taui 0]	
Absolute tolerance:	
auto	
State Name: (e.g., 'position')	
п	
OK Cancel Help Apply	/

PI-control of integrator (level control). G = 1/s, taui=1. VARY Kc

PI-control of integrator (level control). G = 1/s, Kc=1. VARY taui

Poles and zeros

g(s) = n(s)/d(s).Example.

$$g(s) = \frac{12s + 6}{30s^2 + 33s + 3}$$

Standard forms:

1. Time constant form

$$g(s) = k \frac{(T_1 s + 1) \cdots}{(\tau_1 s + 1)(\tau_2 s + 1) \cdots}$$
$$g(s) = 2 \frac{2s + 1}{(10s + 1)(s + 1)}$$

2. Pole-zero form,

p=pole, z=zero

(more general, for unstable and complex poles/zeros)

$$g(s) = c \frac{(s-z_1)\cdots}{(s-p_1)(s-p_2)\cdots}$$

$$g(s) = \frac{12}{30} \frac{s + 0.5}{(s + 0.1)(s + 1)}$$

$$z_1 = -1/T_1 = -1/2 = -0.5$$

$$p_1 = -1/\tau_1 = -1/10 = -0.1;$$

$$p_2 = -1/\tau_2 = -1/1 = -1$$

Poles and zeros

- Transfer functions G(s) of linear, time-invariant systems with time delay are ratios of two polynomials in *s* (Laplace variable)
 - G(s) = n(s)/d(s)
- Polynomials have roots. root in denominator, d(s)=0: $G(s) \rightarrow \infty$ "pole" (x) root in numerator, n(s)=0: $G(s) \rightarrow 0$ "zero" (0)

- Effect on dynamics:
 - Poles determine stability and fast or slow dynamics
 - d(s)=(τs+1). «Stable» LHP-poles slow down the response - slower for large τ
 - Poles in right half plane (RHP): Unstable .
 - Example: g(s)=1/(s-1). Has RHP-pole at s=1
 - Complex poles (=eigenvalues of A): Oscillations with frequency ω
 - Example: $g(s) = 1/(s^2 + s + 1)$. Solve $d(s) = s^2 + s + 1 = 0$. Get poles $s1 = -0.5 + 0.87^*i$, $s2 = -0.5 0.87^*i$, $\omega = 0.87^*i$
 - Zeros are responsible for shape of response
 - n(s)=(Ts+1). Zeros in left half plane (LHP): «Lift» the response
 - give overshoot for large T.
 - n(s)=(-Ts+1). Zeros in right half plane (RHP): always give inverse response
 - Inverse response makes problems for feedback control
 - Example: g(s)=(-0.5s+1) / (10s^2+11s+1). Has RHP-zero at s=2

- Zeros are common in practise
- Occur when there are several «paths» to the output.
- RHP zero: «competing effects where slow wins (has largest gain)»

• Example 1 $g_1(s) = \frac{2}{10s+1}$, $g_2(s) = \frac{0.3}{s+1}$ $g(s) = g_1 + g_2 = \frac{2(s+1)+0.3(10s+1)}{(10s+1)(s+1)} = 2.3 \frac{2.17s+1}{(10s+1)(s+1)}$ All coefficients positive: LHP zero

• Example 2
$$g_1(s) = \frac{2}{10s+1}$$
, $g_2(s) = -\frac{0.3}{s+1}$
 $g(s) = g_1 + g_2 = \frac{2(s+1) - 0.3(10s+1)}{(10s+1)(s+1)} = 1.7 \frac{4}{(10s+1)(s+1)}$ Sign change: RHP zero) Inverse response

• Example 3

$$g_{1}(s) = -\frac{0.3}{10s+1}, \quad g_{2}(s) = \frac{2}{s+1}$$

$$g(s) = g_{1} + g_{2} = \frac{2^{10}(s+1) - 0.3(10(s+1))}{(10s+1)(s+1)} = 1.7 \frac{11.3s+1}{(10s+1)(s+1)}$$
Note; Overshoot since 11.3>10
(overshoot: competing effects where fast wins)

5

10

15 20 25 30 35 40 Time (seconds) Start here week 7

Figure 5.3 Step response of an overdamped secondorder system (Eq. 5-14) for different values of τ_a ($\tau_1 = 4$, $\tau_2 = 1$).

Summary poles and zeros

- $G(s) = n(s) / d(s) = k'(s-z_1) / (s-p_1)(s-p_2)..$
- Example: $G(s) = 4 (3s-1)/(s^2+s-2)$, Get: k'=12, $z_1=1/3$, $p_1=-2 p_2=1$
- Poles p (=eigenvalues of A-matrix)
 - Determine speed of response, exp(p*t)
 - Negative sign in d(s)) p_2 in RHP: unstable, exp(p_2 *t) ! 1 (NEED control)
 - Pole p complex: oscillating response
- Zeros z
 - Determine shape of response
 - Negative sign in n(s)) z_1 in RHP: inverse response (BAD for control)
 - LHP-zero may give overshoot

Approximation of time delay

- Time delay is a bit difficult because it's «infinite order»
 - So cannot directly use with g(s)=n(s)/d(s) (where n and d are polynomials)
 - or with «state-space» form (dx/dt=Ax+Bu)
- Taylor expansion of e^x (x=-θs)

$$e^x = \sum_{n=0}^\infty rac{x^n}{n!} = 1 + x + rac{x^2}{2!} + rac{x^3}{3!} + rac{x^4}{4!} + \cdots$$

• Approximation of delay as n(s)/d(s):

$$e^{-\theta s} = 1 - \theta s + \dots \approx -\theta s + 1$$

RHP-zero approximation (we use this for SIMC)

$$e^{-\theta s} = \frac{1}{e^{\theta s}} \approx \frac{1}{\theta s + 1}$$
$$e^{-\theta s} = \frac{e^{-\frac{\theta}{2}s}}{e^{-\frac{\theta}{2}s}} \approx \frac{-\frac{\theta}{2}s + 1}{\frac{\theta}{2}s + 1}$$
$$e^{-\theta s} \approx \frac{(-\frac{\theta}{2n}s + 1)^n}{(\frac{\theta}{2n}s + 1)^n}$$

1st order approximation

Pade approximation (better)

n'th order Pade approximation (even better)

Simple approximations of time delay

Example: Step response of first-order system plus delay

n'th order Pade approximation

Accurate for large n

$$e^{-\theta s} \approx \frac{(-\frac{\theta}{2n}s+1)^n}{(\frac{\theta}{2n}s+1)^n}$$

Note: Number of RHP-zeros = number of 0-crossings of step response

s=tf('s') theta=1 g0 =exp(-theta*s) % Original time delay g1 = (-theta*s/2+1)/(theta*s/2+1) % 1st-order Pade-approximation g2 = (-theta*s/4+1)^2/(theta*s/4+1)^2 % 2nd-order Pade-approximation g3 = (-theta*s/6+1)^3/(theta*s/6+1)^3 % 3rd-order Pade-approximation h=1/(s+1) step(g0*h,g1*h,g2*h,g3*h) axis([0 5 -0.2 1.1])

Why use Pade? To get model on state space form, dx/dt=Ax+Bu

Approximations of transfer functions

- Going the «other way»
- Want to approximate g(s)=n(s)/d(s) as first-order plus delay
 - «Skogestad half rule» to find effective delay. IMPORTANT!
 - see slides SiS6 for SIMC-rule

Extra: Examples of dynamic model structures

How do we get zeros?

RHP-zero (inverse response)

Response in y=T to a 10% step increase in u=w_H =0.1: 41.5C 40C 20s

Two effects: 1) Direct effect of mixing: $g_1(s)=15$ 2) Indirect effect of changed T_h : $g_2(s) = -30/(20s+1)$ $g(s) = g_1 + g_2 = -15\frac{-20s+1}{20s+1}$

Model derivation

1. Model. Assume: Mass m [kg] in heater constant c_P constant Energy balance heater + mixer: $\frac{d(mc_PT_h)}{dt} = w_h c_P (T_0 - T_h) + Q$ $T = \frac{w_h T_h + w_e T_e}{w_e + w_h}$ 2. Linearize: $y = \Delta T, x = \Delta T_h, u = \Delta w_h$ $\tau \frac{dx}{dt} = -x + ku$ y = Cx + Du $k = \frac{T_e^* - T_h^*}{w_*^*}$

 $\tau = m/w_h^*$

 $C = \frac{w_h^*}{w_c^* + w_h^*}$

 $D = \frac{T_h^* - T^*}{w_c + w_h^*}$

3. Nominal steady-state data:

$$T_0 = 10C, T_h = 70C, T = 40C$$

 $w_h = w_c = 1kg/s, m = 20kg$
Gives:
 $k = \frac{T_o^* - T_h^*}{w_h^*} = \frac{10 - 70}{1} = -60$
 $\tau = m/w_h^* = 20/1 = 20$
 $C = \frac{w_h^*}{w_c^* + w_h^*} = 0.5$
 $D = \frac{T_h^* - T^*}{w_c + w_h^*} = \frac{70 - 40}{2} = 15$
4. Transfer function:

4. Transfer function: y(s) = G(s)u(s) $G(s) = C \frac{k}{\tau s + 1} + D$ $= 0.5 \frac{-60}{20s + 1} + 15$ $= -15 \frac{-20s + 1}{20s + 1}$

Zero at 0 (no steady-state effect)

Model derivation

- 1. Model. Assume: Mass m [kg] in heater constant c_P constant Energy balance heater + mixer: $\frac{d(mc_PT_h)}{dt} = (1 - \alpha)wc_P(T_0 - T_h) + Q$ $T = (1 - \alpha)T_h + \alpha T_c$
 - 2. Linearize: $y = \Delta T, x = \Delta T, u = \alpha$ $\tau \frac{dx}{dt} = -x + ku$ y = Cx + Du $k = -\frac{T_o^* - T_h^*}{(1 - \alpha^*)}$ $\tau = m/w_h^*$ $C = (1 - \alpha^*)$ $D = (T_o^* - T_h^*)$

- 3. Nominal steady-state data: $T_0 = 10C, T_h = 70C, T = 64C$ $w = 1kg/s, \alpha = 0.1, m = 20kg$ Gives: $k = -\frac{T_o^* - T_h^*}{(1 - \alpha^*)} = -\frac{10 - 70}{0.9} = 66.67$ $\tau = m/w_h^* = 20/0.9 = 22$ $C = (1 - \alpha^*) = 0.9$ $D = (T_o^* - T_h^*) = -60$
- 4. Transfer function:

$$\begin{split} y(s) &= G(s)u(s) \\ G(s) &= C\frac{k}{\tau s + 1} + D \\ &= 0.9\frac{66.67}{22s + 1} - 60 \\ &= 60(\frac{1}{22s + 1} - 1) = -60\frac{22s}{22s + 1} \end{split}$$