

State Space-based MPC using a Kalman Filter

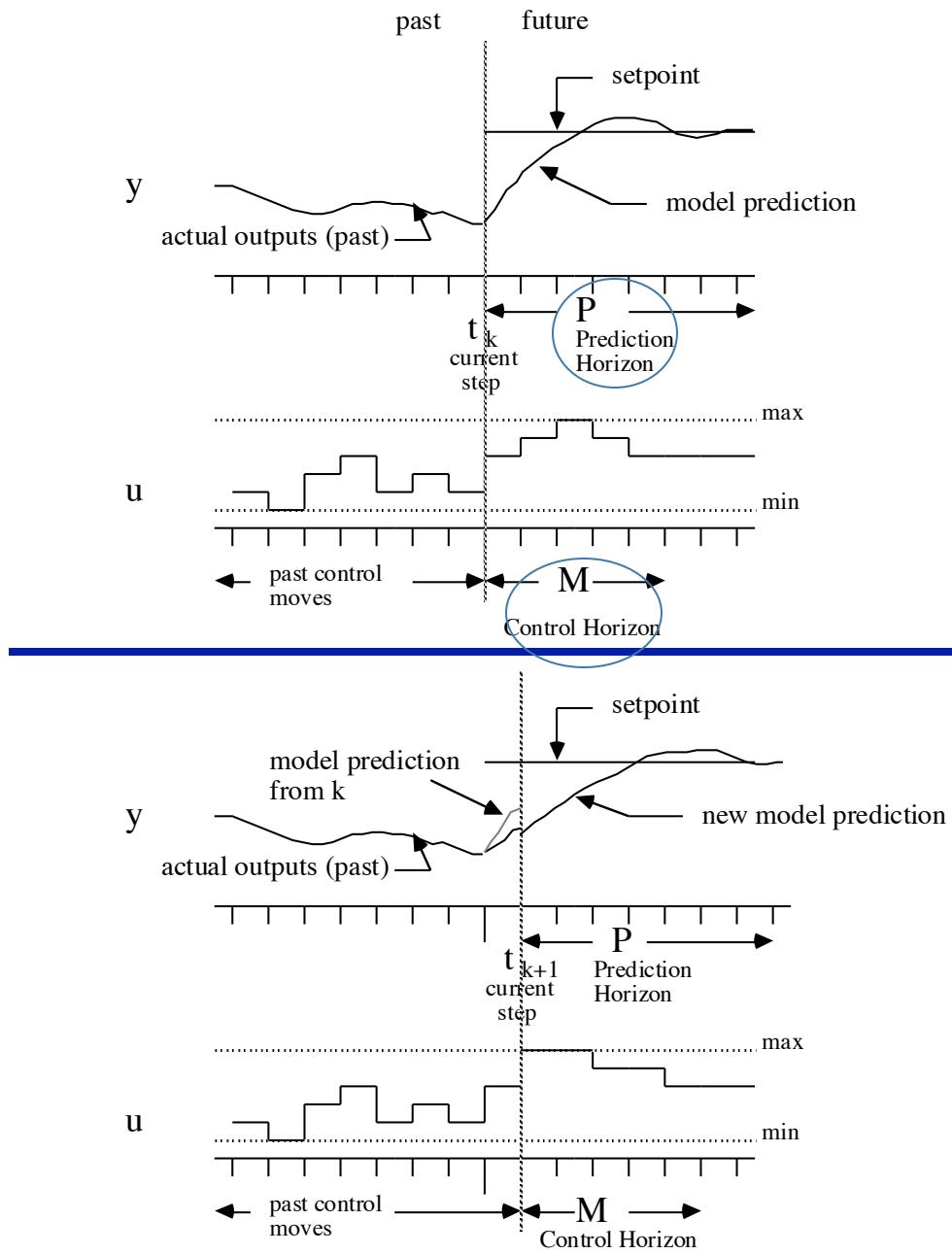
B. Wayne Bequette

- Quick Review
- State Space Models
 - Unconstrained and Constrained Solutions
- Kalman Filter for Disturbance Estimation
- Summary



Rensselaer
Chemical and Biological Engineering

Model Predictive Control (MPC)



- Constraints
 - Multivariable
 - Time-delays
 - Quadratic objective function
 - Quadratic program (QP)
constraints
 - State space model
 - Disturbances
additive output vs. state estimation
- B. Wayne Bequette

Original DMC Approach to Plant-Model Mismatch

$$\hat{x}_{k|k-1} = \Phi \hat{x}_{k-1|k-1} + \Gamma u_{k-1}$$

$$\hat{y}_{k|k-1} = C \hat{x}_{k|k-1}$$

Prediction at step k, based on information at k-1

$$\hat{p}_{k|k} = y_k - \hat{y}_{k|k-1}$$

Model output predicted from k-1

$$\hat{y}_{k|k} = \hat{y}_{k|k-1} + \hat{p}_{k|k}$$

“additive output” disturbance assumption

Forces the model “corrected output” equal to measured output

Notice that Model States are Not “Corrected”

$$\hat{x}_{k|k} = \hat{x}_{k|k-1}$$

Model Prediction to k+1

$$\hat{x}_{k+1|k} = \Phi \hat{x}_{k|k} + \Gamma u_k$$

$$\hat{p}_{k+1|k} = \hat{p}_{k|k}$$

Assumes future corrections
equal to current correction

$$\hat{y}_{k+1|k} = C \hat{x}_{k+1|k} + \hat{p}_{k+1|k} =$$

$$= C \Phi \hat{x}_{k|k} + C \Gamma u_k + \hat{p}_{k+1|k}$$

$$= C \Phi \hat{x}_{k|k} + C \Gamma u_{k-1} + C \Gamma \Delta u_k + \hat{p}_{k+1|k}$$

Continue Output Predictions

$$\underbrace{\begin{bmatrix} \hat{y}_{k+1|k} \\ \hat{y}_{k+2|k} \\ \vdots \\ \hat{y}_{k+P|k} \end{bmatrix}}_{\hat{Y}} = \underbrace{\begin{bmatrix} C\Phi \\ C\Phi^2 \\ \vdots \\ C\Phi^P \end{bmatrix} \hat{x}_{k|k} + \begin{bmatrix} I \\ I \\ \vdots \\ I \end{bmatrix} \hat{p}_{k|k} + \begin{bmatrix} C\Gamma \\ C\Phi\Gamma + C\Gamma \\ \vdots \\ \sum_{i=1}^P C\Phi^{i-1}\Gamma \end{bmatrix} u_{k-1}}_{\text{"free" or "unforced response" (if no more control moves are made)}}$$

$$f \quad \begin{bmatrix} C\Gamma & 0 & \cdots & 0 \\ C\Phi\Gamma + C\Gamma & C\Gamma & 0 & 0 \\ \vdots & \vdots & & \vdots \\ \sum_{i=1}^P C\Phi^{i-1}\Gamma & \sum_{i=1}^{P-1} C\Phi^{i-1}\Gamma & \cdots & \Delta u_{k+M-1} \end{bmatrix} \underbrace{\Delta u_k \\ \Delta u_{k+1} \\ \vdots \\ \Delta u_{k+M-1}}_{\text{"forced" response}} \quad \Delta u_f$$

Output Predictions

$$\underbrace{\begin{bmatrix} \hat{y}_{k+1|k} \\ \hat{y}_{k+2|k} \\ \vdots \\ \hat{y}_{k+P|k} \end{bmatrix}}_{\hat{Y}} = \underbrace{\begin{bmatrix} C\Phi \\ C\Phi^2 \\ \vdots \\ C\Phi^P \end{bmatrix} \hat{x}_{k|k} + \begin{bmatrix} I \\ I \\ \vdots \\ I \end{bmatrix} \hat{p}_{k|k} + \begin{bmatrix} S_1 \\ S_2 \\ \vdots \\ S_P \end{bmatrix} u_{k-1}}_{\text{"free" or "unforced response" (if no more control moves are made)}}$$

\hat{Y}

$$f \left[\begin{array}{cccc} S_1 & 0 & \cdots & 0 \\ S_2 & S_1 & 0 & 0 \\ \vdots & \vdots & & \\ S_P & S_{P-1} & \cdots & S_{P-M+1} \end{array} \right] \left[\begin{array}{c} \Delta u_k \\ \Delta u_{k+1} \\ \vdots \\ \Delta u_{k+M-1} \end{array} \right]$$

"forced" response

S_f

Δu_f

Same Optimization Problem as Before

$$\min_{\Delta u_f} J = \underbrace{\sum_{i=1}^P (r_{k+i|k} - \hat{y}_{k+i|k})^T W^y (r_{k+i|k} - \hat{y}_{k+i|k})}_{\hat{E}^T W^Y \hat{E}} + \underbrace{\sum_{i=0}^{M-1} \Delta u_{k+i}^T W^u \Delta u_{k+i}}_{\Delta u_f^T W^U \Delta u_f}$$

Where

$$W^Y = \begin{bmatrix} W^y & 0 & 0 \\ 0 & \ddots & 0 \\ 0 & 0 & W^y \end{bmatrix} \quad W^U = \begin{bmatrix} W^u & 0 & 0 \\ 0 & \ddots & 0 \\ 0 & 0 & W^u \end{bmatrix}$$

future setpoints

and

$$\hat{E} = r - \hat{Y} = \underbrace{r - f}_{E} - S_f \Delta u_f$$

“unforced” error

so

$$\hat{E} = E - S_f \Delta u_f$$

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Optimization Problem...

$$\begin{aligned}\hat{E}^T W^Y \hat{E} &= (E - S_f \Delta u_f)^T W^Y (E - S_f \Delta u_f) \\ &= E^T W^Y E - 2\Delta u_f^T S_f^T W^Y E + \Delta u_f^T S_f^T W^Y S_f \Delta u_f\end{aligned}$$

so

$$\min_{\Delta u_f} J = \hat{E}^T W^Y \hat{E} + \Delta u_f^T W^U \Delta u_f$$

can be written

$$\min_{\Delta u_f} J = E^T W^Y E + \Delta u_f^T (S_f^T W^Y S_f + W^U) \Delta u_f - 2\Delta u_f^T S_f^T W^Y E$$

and the unconstrained solution is found from

$$\partial J / \partial \Delta u_f = 0$$

Unconstrained Solution

Analytical Solution for Unconstrained System

$$\Delta u_f = \left(S_f^T W^Y S_f + W^U \right)^{-1} S_f^T W^Y E$$

“unforced” error 

In practice, do not actually invert a matrix. Solve as set of simultaneous equations (or use \ in MATLAB)

$$\Delta u_f = \left(S_f^T W^Y S_f + W^U \right) \backslash S_f^T W^Y E$$

Vector of Control Moves

$$\Delta u_f = \begin{bmatrix} \Delta u_k \\ \Delta u_{k+1} \\ \vdots \\ \Delta u_{k+M-1} \end{bmatrix}$$

current and future moves

Although a set of control moves is computed, only the first move Δu_k is implemented. The next output at $k+1$ is obtained, then a new optimization problem is solved.

Problems with “Classical MPC” (e.g. DMC)

- Finite Step/Impulse Models Limited
 - Limited to open-loop stable processes (there is no corrective feedback to model states)
- Additive Output Disturbance Assumption
 - Poor performance for input step disturbances
 - No explicit measurement noise trade-off

Ind. Eng. Chem. Res. 2002, 41, 3745–3750

Process Control: As Taught vs as Practiced

Francis G. Shinskey



LIMITATIONS OF DYNAMIC MATRIX CONTROL

P. LUNDSTRÖM,^{1,†} J. H. LEE^{1,‡}, M. MORARI^{1,§} and S. SKOGESTAD²

Computers chem. Engng Vol. 19, No. 4, pp. 409–421, 1995

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MPC, Including Disturbance Models

$$\begin{aligned} \underset{(nx1)}{x_{k+1}} &= \underset{(nxn)(nx1)}{\Phi x_k} + \underset{(nxn_u)(n_u x1)}{\Gamma u_k} + \underset{(nxn_d)}{\Gamma^d} \underset{(n_d x1)}{d_k} \\ \underset{(n_y x1)}{y_k} &= \underset{(n_y x n)(nx1)}{C x_k} \end{aligned}$$

For perturbations
to manipulated
inputs
 $\overbrace{\Gamma^d} = \Gamma$

with known current state, easy to propagate estimates

$$x_{k+1} = \Phi x_k + \Gamma u_k + \Gamma^d d_k$$

$$y_{k+1} = C x_{k+1} = C \Phi x_k + C \Gamma u_k + C \Gamma^d d_k$$

and, using control changes

$$u_k = u_{k-1} + \Delta u_k$$

$$y_{k+1} = C \Phi x_k + C \Gamma u_{k-1} + C \Gamma^d d_k + C \Gamma \Delta u_k$$

Next, use estimated disturbance

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Based on an Estimated Disturbance

Predicted output	State estimate	Previous input	Estimated disturbance	Current Input change
$\hat{y}_{k+1 k}$	$C\Phi\hat{x}_{k k}$	$C\Gamma u_{k-1}$	$C\Gamma^d \hat{d}_{k k}$	$C\Gamma\Delta u_k$

$$\hat{y}_{k+1|k} = C\Phi\hat{x}_{k|k} + C\Gamma u_{k-1} + C\Gamma^d \hat{d}_{k|k} + C\Gamma\Delta u_k$$

Estimate at k+1
Measurements through k
Estimate at k
Measurements through k
Estimated disturbance
Current Input change

Now, propagate the prediction for P steps into the future

Output Predictions

$$\begin{aligned}
 \begin{bmatrix} \hat{y}_{k+1|k} \\ \hat{y}_{k+2|k} \\ \vdots \\ \hat{y}_{k+P|k} \end{bmatrix} &= \underbrace{\begin{bmatrix} C\Phi \\ C\Phi^2 \\ \vdots \\ C\Phi^P \end{bmatrix} \hat{x}_{k|k}}_{\hat{Y}} + \underbrace{\begin{bmatrix} C\Gamma \\ C\Phi\Gamma + C\Gamma \\ \vdots \\ \sum_{i=1}^P C\Phi^{i-1}\Gamma \end{bmatrix} u_{k-1}}_{\text{"free" or "unforced response" (if no more control moves are made)}} + \underbrace{\begin{bmatrix} C\Gamma^d \\ C\Phi\Gamma^d + C\Gamma^d \\ \vdots \\ \sum_{i=1}^P C\Phi^{i-1}\Gamma^d \end{bmatrix} \hat{d}_{k|k}}_{\text{Estimated disturbance (constant in future)}}
 \end{aligned}$$

$$f \begin{bmatrix} C\Gamma & 0 & \cdots & 0 \\ C\Phi\Gamma + C\Gamma & C\Gamma & 0 & 0 \\ \vdots & \vdots & & \Delta u_k \\ \sum_{i=1}^P C\Phi^{i-1}\Gamma & \sum_{i=1}^{P-1} C\Phi^{i-1}\Gamma & \cdots & \Delta u_{k+1} \\ & & & \vdots \\ & & & \Delta u_{k+M-1} \end{bmatrix} \Delta u_f$$

"forced" response

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Output Predictions (using S notation)

$$\underbrace{\begin{bmatrix} \hat{y}_{k+1|k} \\ \hat{y}_{k+2|k} \\ \vdots \\ \hat{y}_{k+P|k} \end{bmatrix}}_{\hat{Y}} = \underbrace{\begin{bmatrix} C\Phi \\ C\Phi^2 \\ \vdots \\ C\Phi^P \end{bmatrix}}_{\text{"free" or "unforced response" (if no more control moves are made)}} \hat{x}_{k|k} + \underbrace{\begin{bmatrix} S_1 \\ S_2 \\ \vdots \\ S_P \end{bmatrix}}_{u_{k-1}} u_{k-1} + \underbrace{\begin{bmatrix} S_1^d \\ S_2^d \\ \vdots \\ S_P^d \end{bmatrix}}_{\hat{d}_{k|k}} \hat{d}_{k|k}$$

Estimated disturbance
 (constant in future)

$$f \quad \begin{bmatrix} S_1 & 0 & \cdots & 0 \\ S_2 & S_2 & 0 & 0 \\ \vdots & \vdots & & \\ S_P & S_{P-1} & \cdots & S_{P-M+1} \end{bmatrix} \begin{bmatrix} \Delta u_k \\ \Delta u_{k+1} \\ \vdots \\ \Delta u_{k+M-1} \end{bmatrix}$$

"forced" response

Δu_f

S_f

Same Optimization Problem as Before

$$\min_{\Delta u_f} J = \underbrace{\sum_{i=1}^P (r_{k+i|k} - \hat{y}_{k+i|k})^T W^y (r_{k+i|k} - \hat{y}_{k+i|k})}_{\hat{E}^T W^Y \hat{E}} + \underbrace{\sum_{i=0}^{M-1} \Delta u_{k+i}^T W^u \Delta u_{k+i}}_{\Delta u_f^T W^U \Delta u_f}$$

Where

$$W^Y = \begin{bmatrix} W^y & 0 & 0 \\ 0 & \ddots & 0 \\ 0 & 0 & W^y \end{bmatrix} \quad W^U = \begin{bmatrix} W^u & 0 & 0 \\ 0 & \ddots & 0 \\ 0 & 0 & W^u \end{bmatrix}$$

future setpoints

and

$$\hat{E} = r - \hat{Y} = \underbrace{r - f}_{E} - S_f \Delta u_f$$

“unforced” error

so

$$\hat{E} = E - S_f \Delta u_f$$

Optimization Problem...

$$\begin{aligned}\hat{E}^T W^Y \hat{E} &= (E - S_f \Delta u_f)^T W^Y (E - S_f \Delta u_f) \\ &= E^T W^Y E - 2\Delta u_f^T S_f^T W^Y E + \Delta u_f^T S_f^T W^Y S_f \Delta u_f\end{aligned}$$

so

$$\min_{\Delta u_f} J = \hat{E}^T W^Y \hat{E} + \Delta u_f^T W^U \Delta u_f$$

can be written

$$\min_{\Delta u_f} J = E^T W^Y E + \Delta u_f^T (S_f^T W^Y S_f + W^U) \Delta u_f - 2\Delta u_f^T S_f^T W^Y E$$

and the unconstrained solution is found from

$$\partial J / \partial \Delta u_f = 0$$

Unconstrained Solution

Analytical Solution for Unconstrained System

$$\Delta u_f = \left(S_f^T W^Y S_f + W^U \right)^{-1} S_f^T W^Y E$$

“unforced” error 

In practice, do not actually invert a matrix. Solve as set of simultaneous equations (or use \ in MATLAB)

$$\Delta u_f = \left(S_f^T W^Y S_f + W^U \right) \backslash S_f^T W^Y E$$

How do we estimate the disturbance?

Concise Review of Optimal Estimation (Kalman Filtering)

- Kalman & Bucy, 1960

Kalman, R. E. (1960b). A new approach to linear filtering and prediction problems. *Transactions of ASME, Journal of Basic Engineering*, 87, 35–45.



Rudy Kalman

- Developed for well-modeled systems with state input and measurement noise
- Success in Aerospace and other applications
- Problems with initial chemical process applications
 - Limited understanding of limitations
- Bias due to model uncertainty
- Need an “appended state” formulation

Stochastic Models

$$x_{k+1} = \Phi x_k + \Gamma u_k + \Gamma^w w_k$$

Input noise
 $\text{cov}(w) = Q$

$$y_k = C x_k + v_k$$

Measurement noise
 $\text{cov}(v) = R$

$$\text{cov}(x_0) = P_0$$

Initial State Uncertainty

Optimal State Estimate

$$J(\hat{y}) = \sum_{i=1}^k (y_i - C\hat{x})^T R^{-1} (y_i - C\hat{x})$$

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State Estimation Procedure

$$\hat{x}_{k|k-1} = \Phi \hat{x}_{k-1|k-1} + \Gamma u_{k-1} \quad \text{Prediction}$$

Estimate at k, based on measurements through k-1

$$\hat{x}_{k|k} = \hat{x}_{k|k-1} + L_k (y_k - C \hat{x}_{k|k-1}) \quad \text{Correction}$$

Estimate at k, based on measurements through k

$$P_k = \Phi P_{k-1} \Phi^T + \Gamma^w Q \Gamma^{wT} - \Phi P_{k-1} C^T (C P_{k-1} C^T + R)^{-1} C P_{k-1} \Phi^T$$

State error covariance estimate

Kalman gain

$$L_k = P_k C^T (C P_k C^T + R)^{-1}$$

Discuss steady-state Kalman gain

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State Covariance Comparison

“Open-loop”

$$P_k = \Phi P_{k-1} \Phi^T + \Gamma^w Q \Gamma^{wT}$$

With measurement (Kalman Filter formulation)

$$P_k = \Phi P_{k-1} \Phi^T + \Gamma^w Q \Gamma^{wT} - \Phi P_{k-1} C^T (C P_{k-1} C^T + R)^{-1} C P_{k-1} \Phi^T$$



Measurement update reduces state estimate covariance

Kalman Filter w/Augmented States

$$\begin{bmatrix} x_{k+1} \\ d_{k+1} \end{bmatrix} = \underbrace{\begin{bmatrix} \Phi & \Gamma^d \\ 0 & I \end{bmatrix}}_{\Phi^a} \underbrace{\begin{bmatrix} x_k \\ d_k \end{bmatrix}}_{x_k^a} + \underbrace{\begin{bmatrix} \Gamma \\ 0 \end{bmatrix}}_{\Gamma^a} u_k + \underbrace{\begin{bmatrix} 0 \\ I \end{bmatrix}}_{\Gamma^{a,w}} w_k$$

process noise
 $\text{cov}(w) = Q$

$$y_k = \underbrace{\begin{bmatrix} C & 0 \end{bmatrix}}_{C^a} \underbrace{\begin{bmatrix} x_k \\ d_k \end{bmatrix}}_{x_k^a} + v_k$$

Augmented state (includes disturbance)
Measurement noise
 $\text{cov}(v) = R$

Predictor-corrector equations:

$$\hat{x}_{k|k-1}^a = \Phi^a \hat{x}_{k-1|k-1}^a + \Gamma^a u_{k-1}$$

Manipulated input

$$\hat{x}_{k|k}^a = \hat{x}_{k|k-1}^a + L_k (y_k - C^a \hat{x}_{k|k-1}^a)$$

Aug. state estimate
Kalman gain
Measured output

Disturbance Portion of Augmented State

$$\hat{x}_{k|k}^a = \begin{bmatrix} \hat{x}_{k|k} \\ \hat{d}_{k|k} \end{bmatrix}$$

Constant Disturbance Into the Future

$$\hat{d}_{k+j|k} = \hat{d}_{k|k}$$

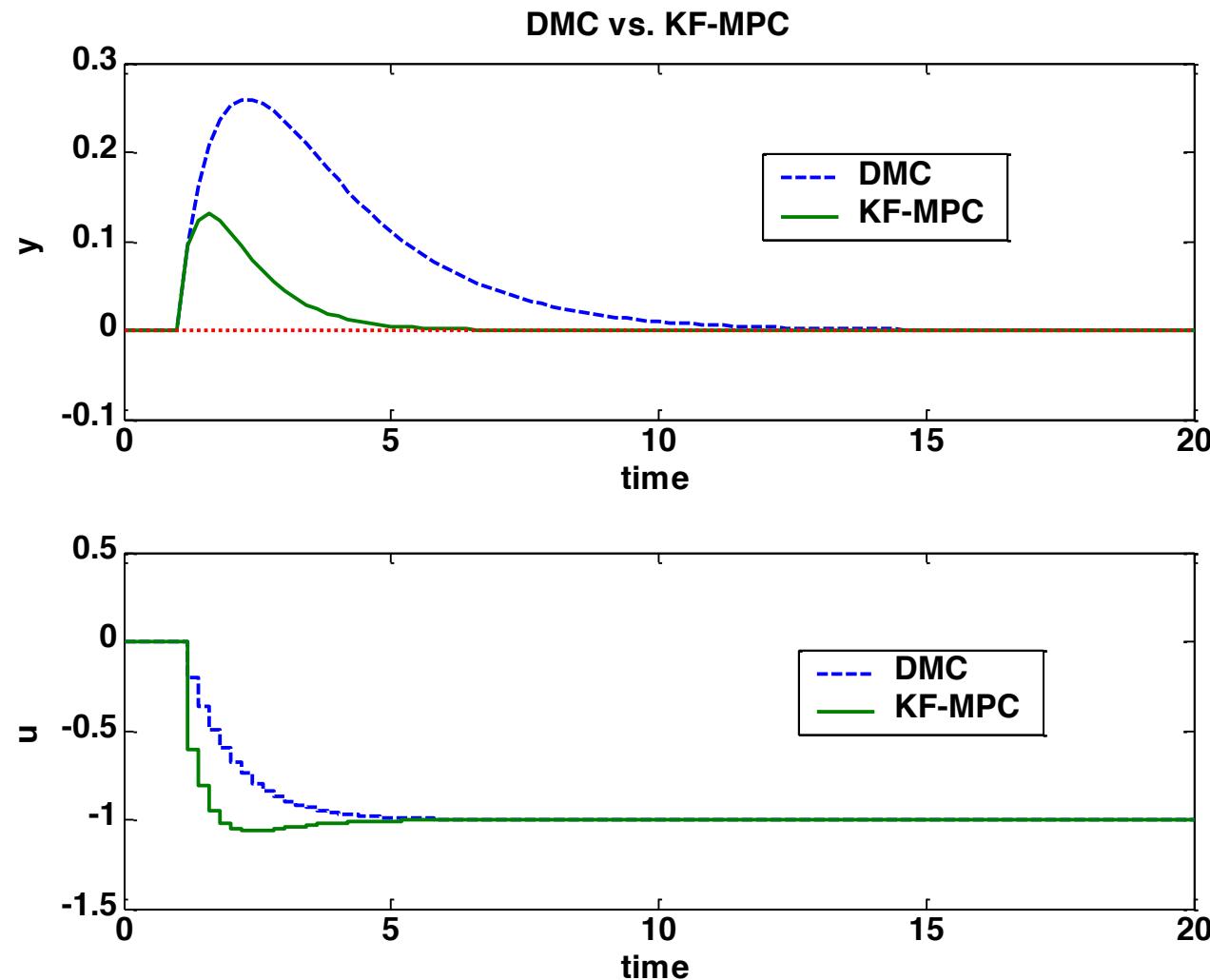
Can use physical knowledge to make better assumptions (periodic effects, etc.)

Example

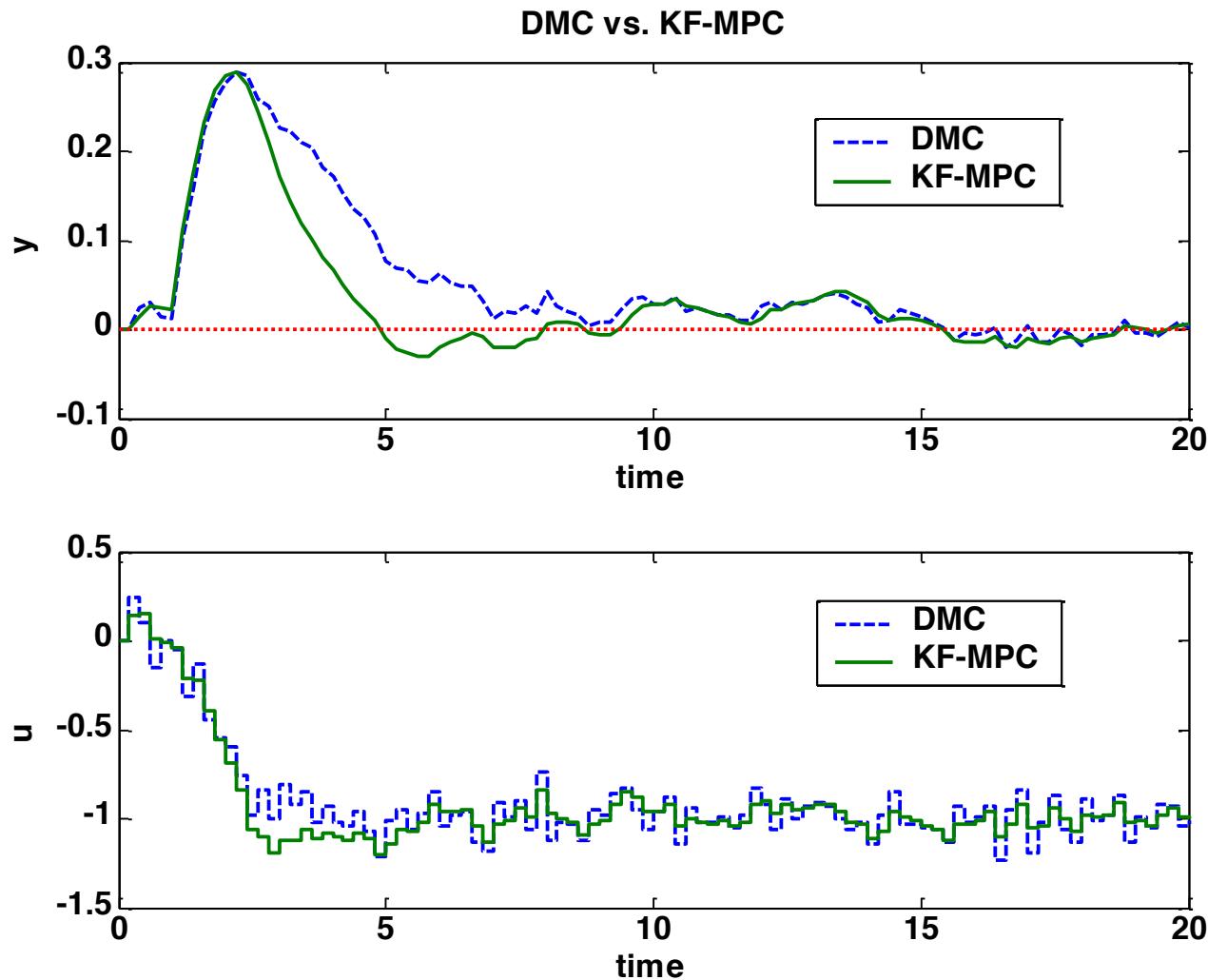
$$g_p(s) = \frac{1}{10s + 1}$$

**Comparison of DMC and KF-based MPC for
a step input disturbance**

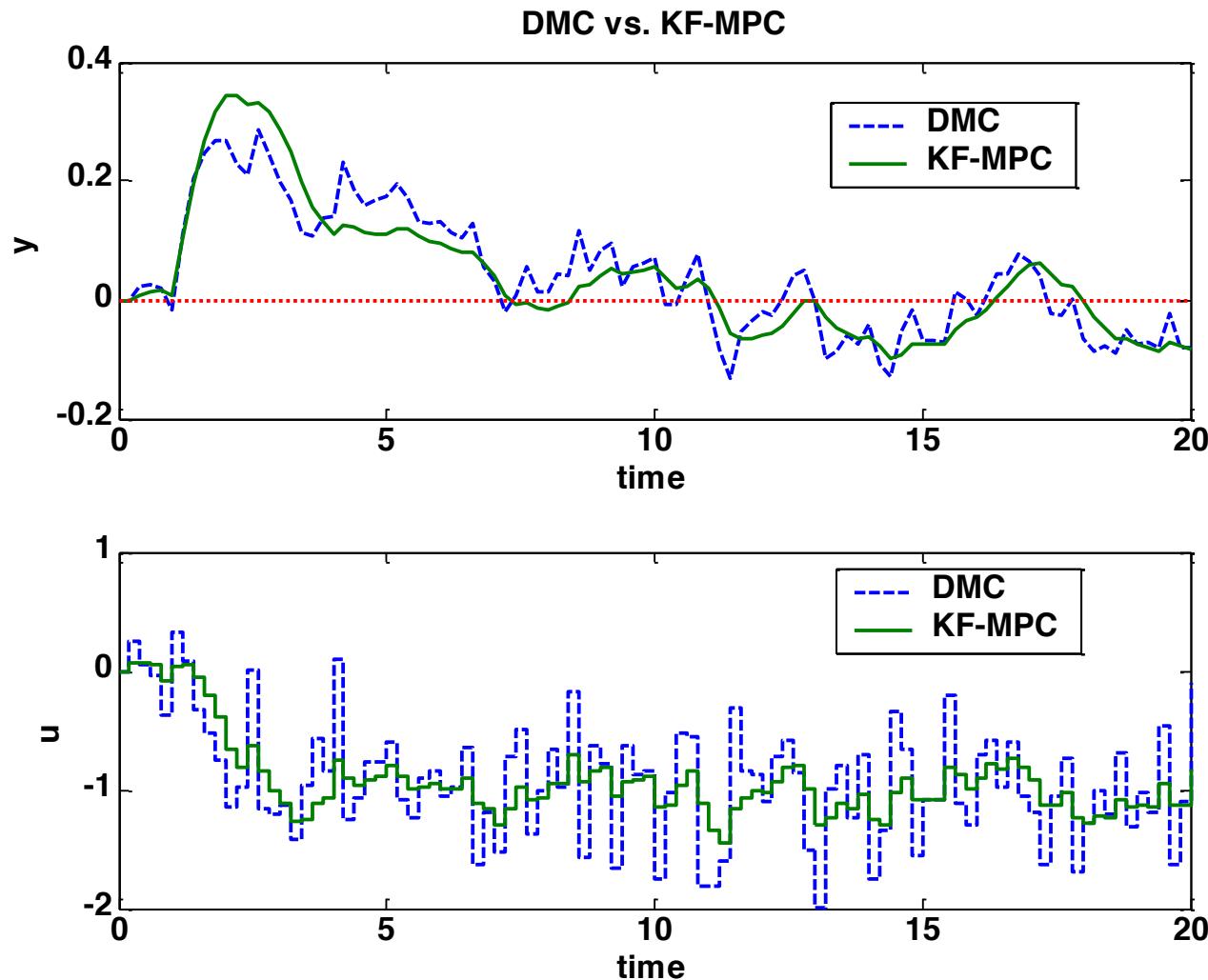
Step Input Disturbance



Measurement Noise



Substantial Measurement Noise



Constrained MPC

Input constraints

$$\begin{array}{c} \mathcal{U}_{\min} \leq \mathcal{U}_{k+i} \leq \mathcal{U}_{\max} \\ (\mathbf{n}_u \times 1) \quad (\mathbf{n}_u \times 1) \quad (\mathbf{n}_u \times 1) \end{array}$$

where $\mathcal{U}_k = \mathcal{U}_{k-1} + \Delta\mathcal{U}_k$

$$\mathcal{U}_{k+1} = \mathcal{U}_{k-1} + \Delta\mathcal{U}_k + \Delta\mathcal{U}_{k+1} \quad \text{etc.}$$

Input “velocity” constraints

$$\Delta\mathcal{U}_{\min} \leq \Delta\mathcal{U}_{k+i} \leq \Delta\mathcal{U}_{\max}$$

Output constraints

$$y_{\min} \leq \hat{y}_{k+i|k} \leq y_{\max}$$

MPC: Constraint Formulation

Input

$$\begin{bmatrix} u_{\min} \\ u_{\min} \\ \vdots \\ u_{\min} \end{bmatrix} \leq \begin{bmatrix} u_{k-1} \\ u_{k-1} \\ \vdots \\ u_{k-1} \end{bmatrix} + \begin{bmatrix} I & 0 & 0 & \cdots & 0 \\ I & I & 0 & \cdots & 0 \\ \vdots & \vdots & & & \\ I & I & I & \cdots & I \end{bmatrix} \begin{bmatrix} \Delta u_k \\ \Delta u_{k+1} \\ \vdots \\ \Delta u_{k+M-1} \end{bmatrix} \leq \begin{bmatrix} u_{\max} \\ u_{\max} \\ \vdots \\ u_{\max} \end{bmatrix}$$

Velocity

$$\begin{bmatrix} \Delta u_{\min} \\ \Delta u_{\min} \\ M \\ \Delta u_{\min} \end{bmatrix} \leq \begin{bmatrix} \Delta u_k \\ \Delta u_{k+1} \\ M \\ \Delta u_{k+M-1} \end{bmatrix} \leq \begin{bmatrix} \Delta u_{\max} \\ \Delta u_{\max} \\ M \\ \Delta u_{\max} \end{bmatrix}$$

Output constraints

$$y_{\min} - f \leq S_f \Delta u_f \leq y_{\max} - f$$

“Free” or “unforced” response

$$f = \underbrace{\begin{bmatrix} C\Phi \\ C\Phi^2 \\ \vdots \\ C\Phi^P \end{bmatrix} \hat{x}_{k|k} + \begin{bmatrix} C\Gamma \\ C\Phi\Gamma + C\Gamma \\ \vdots \\ \sum_{i=1}^P C\Phi^{i-1}\Gamma \end{bmatrix} u_{k-1}}_{\text{"free" or "unforced response" (if no more control moves are made)}} + \begin{bmatrix} C\Gamma^d \\ C\Phi\Gamma^d + C\Gamma^d \\ \vdots \\ \sum_{i=1}^P C\Phi^{i-1}\Gamma^d \end{bmatrix} \hat{d}_{k|k}$$

QP Codes: “one-sided” form

$$\underbrace{\begin{bmatrix} I & 0 & 0 & \cdots & 0 \\ I & I & 0 & \cdots & 0 \\ \vdots & \vdots & & & \\ I & I & I & \cdots & I \end{bmatrix}}_{A_1} \underbrace{\begin{bmatrix} \Delta u_k \\ \Delta u_{k+1} \\ \vdots \\ \Delta u_{k+M-1} \end{bmatrix}}_{\Delta u_f} \leq \underbrace{\begin{bmatrix} u_{\min} - u_{k-1} \\ u_{\min} - u_{k-1} \\ \vdots \\ u_{\min} - u_{k-1} \end{bmatrix}}_{b_1}$$

$$- \underbrace{\begin{bmatrix} I & 0 & 0 & \cdots & 0 \\ I & I & 0 & \cdots & 0 \\ \vdots & \vdots & & & \\ I & I & I & \cdots & I \end{bmatrix}}_{A_2} \underbrace{\begin{bmatrix} \Delta u_k \\ \Delta u_{k+1} \\ \vdots \\ \Delta u_{k+M-1} \end{bmatrix}}_{\Delta u_f} \leq \underbrace{\begin{bmatrix} u_{k-1} - u_{\max} \\ u_{k-1} - u_{\max} \\ \vdots \\ u_{k-1} - u_{\max} \end{bmatrix}}_{b_2}$$

“one-sided” form for input constraints

$$\begin{bmatrix} A_1 \\ A_2 \end{bmatrix} \Delta u_f \geq \begin{bmatrix} b_1 \\ b_2 \end{bmatrix}$$

“one-sided” form for output constraints

$$S_f \Delta u_f \geq y_{\min} - f$$

$$\underbrace{-S_f \Delta u_f \geq -y_{\max} + f}_{\text{---}}$$

$$\begin{bmatrix} A_3 \\ A_4 \end{bmatrix} \Delta u_f \geq \begin{bmatrix} b_3 \\ b_4 \end{bmatrix}$$

Combining all constraints

$$\underbrace{\begin{bmatrix} A_1 \\ A_2 \\ A_3 \\ A_4 \end{bmatrix} \Delta u_f \geq \begin{bmatrix} b_1 \\ b_2 \\ b_3 \\ b_4 \end{bmatrix}}_{\text{---}}$$

$$A \Delta u_f \geq b$$

Quadratic Program (QP)

Objective Function

$$\min_{\Delta u_f} \Phi = \frac{1}{2} \cdot \Delta u_f^T H \Delta u_f + c^T \Delta u_f$$

Decision variables

s.t.

$$A \Delta u_f \geq b$$

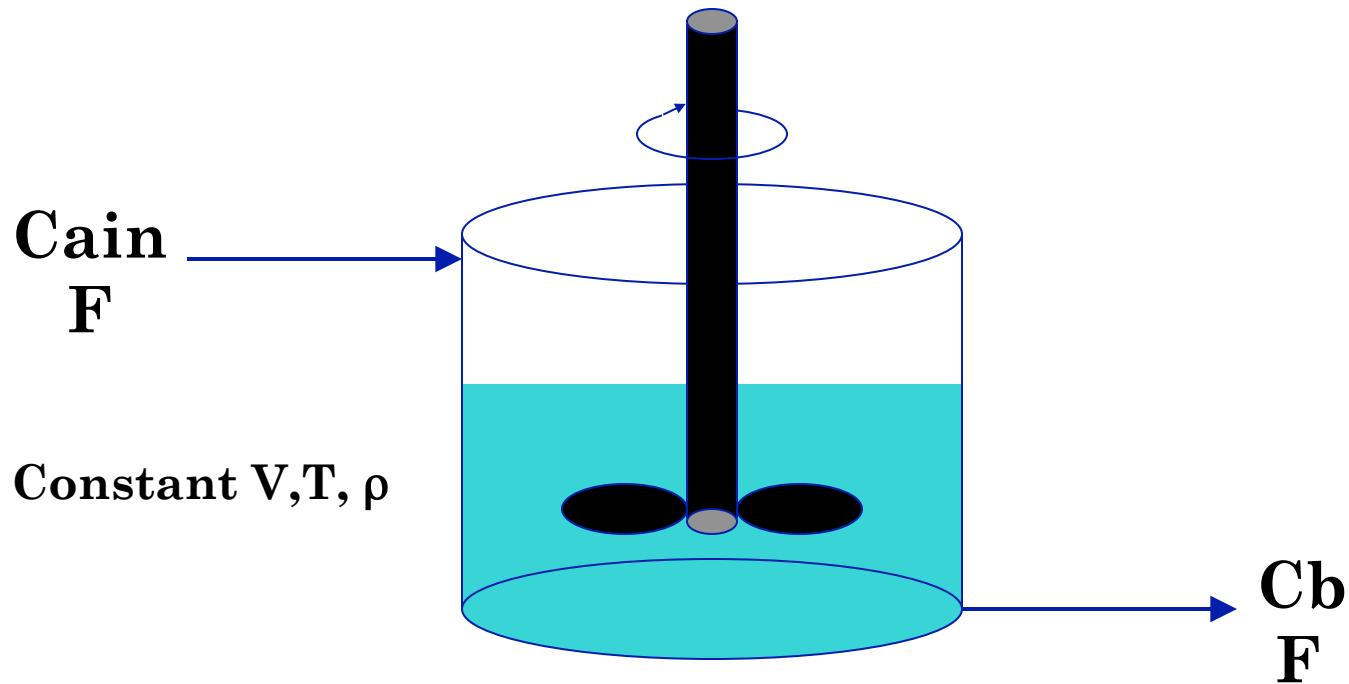
$$\Delta u_{\min} \leq \Delta u_f \leq \Delta u_{\max}$$

Where

$$H = S_f^T W^Y S_f + W^U$$

$$c = -S_f^T W^Y E$$

Inverse Response Process: Van de Vusse Reaction

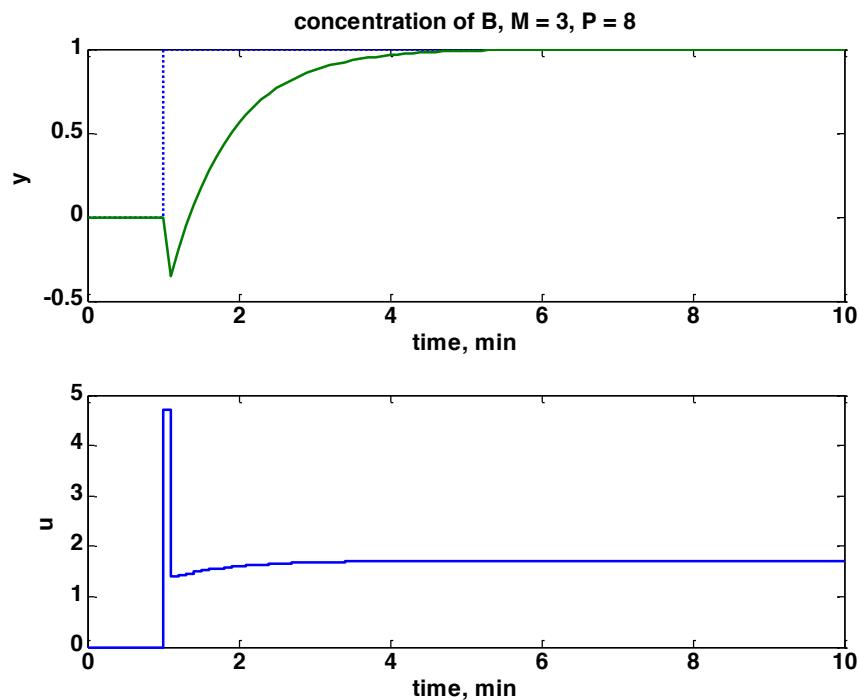


$$\frac{dC_a}{dt} = -k_1 C_a - k_3 C_a^2 + (C_{ain} - C_a)u$$

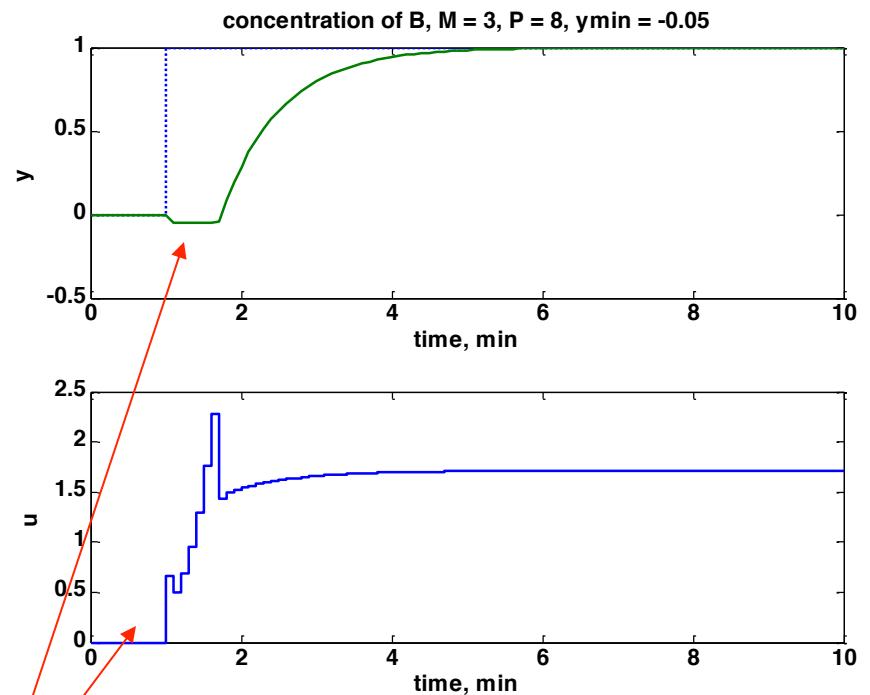
$$\frac{dC_b}{dt} = k_1 C_a - k_2 C_b - C_b u \quad \text{where } u = F/V$$

Output Constraints: Inverse Response Example

Unconstrained



$y \geq -0.05$

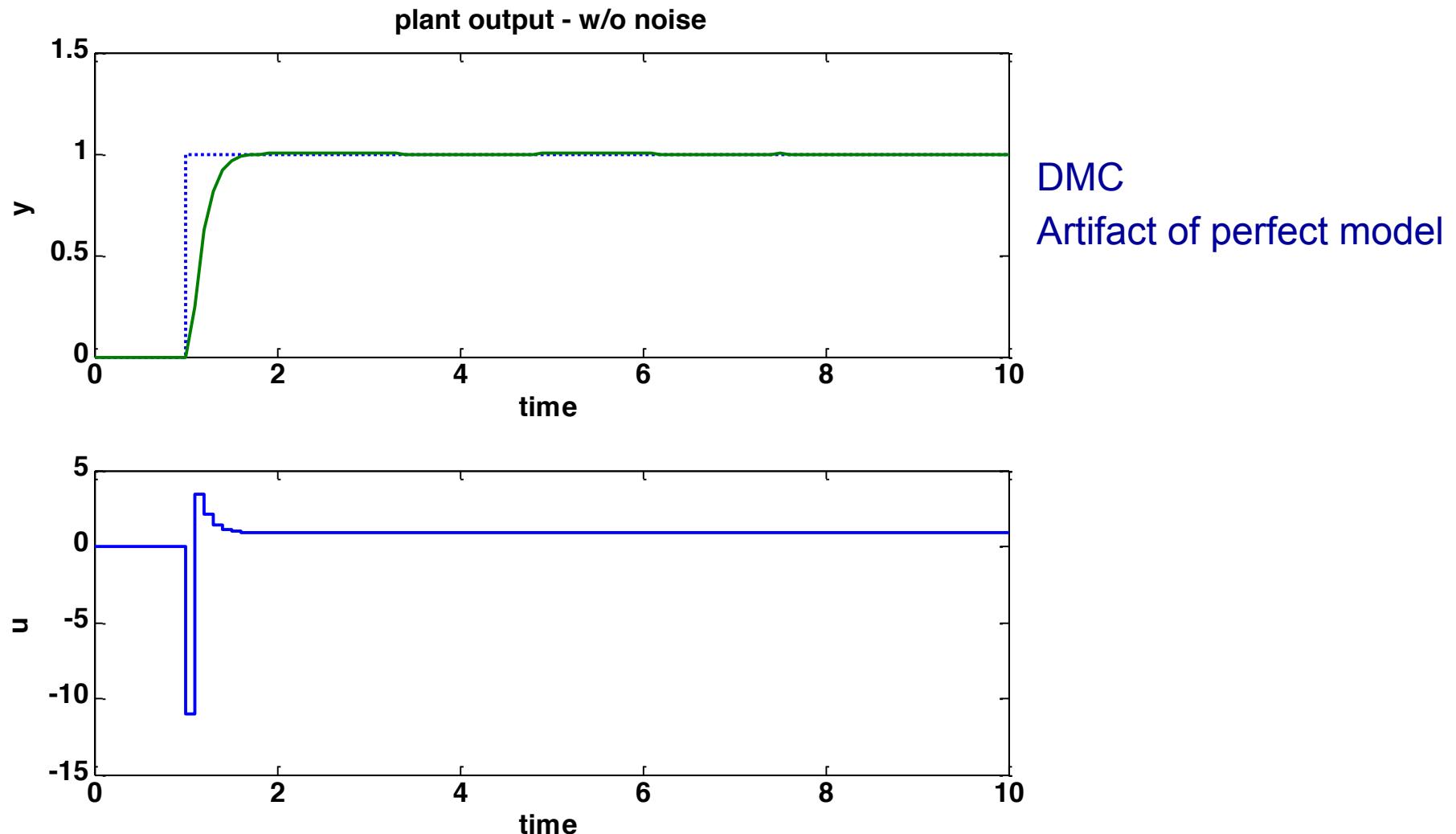


Less control action in order to meet output constraint

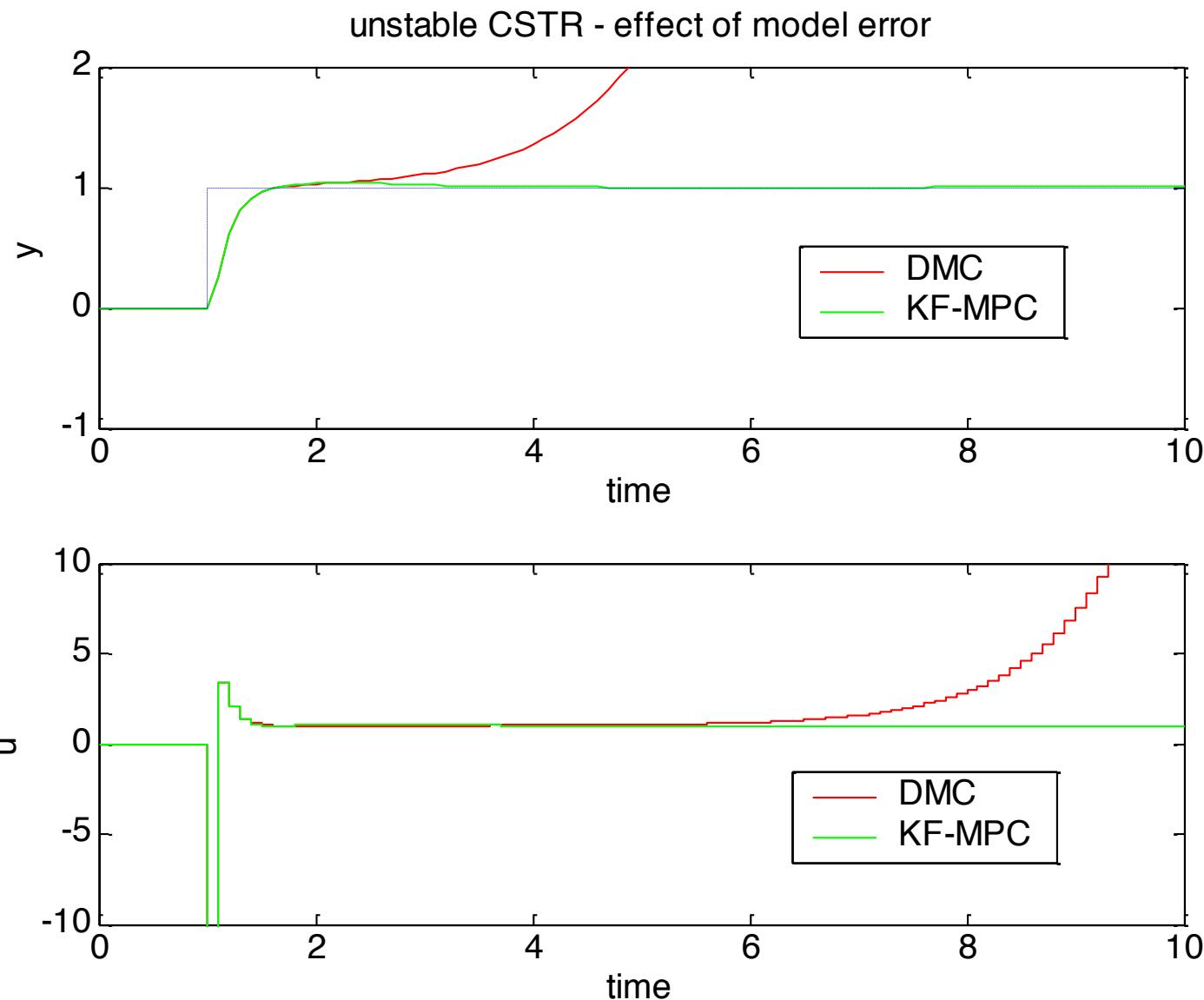
Output Constraints

- May not exist a feasible solution
- Relax constraints until feasible
- “Soft constraints” – Penalize in Objective Function

Unstable CSTR Example



Effect of Model Error



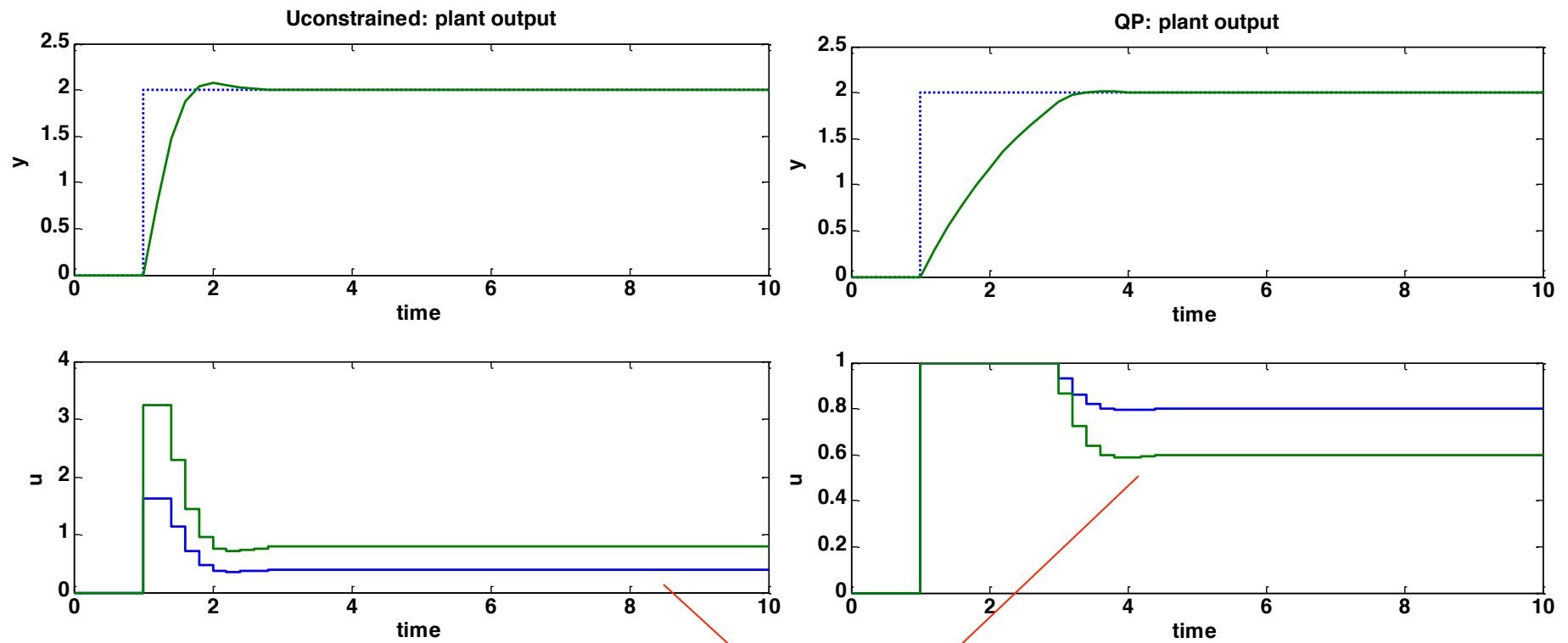
Multivariable Examples

- Have shown clear advantages of KF-based MPC over DMC for SISO
 - Disturbance rejection
 - Unstable processes
- In general, MPC is especially powerful for handling constrained, multivariable systems (including “nonsquare”)
- Two “simple” example systems follow
 - Two-input, one-output (parallel valves)
 - Quadruple tank (two-input, two-output)

2 input – 1 output

$$y(s) = \frac{1}{2s+1}u_1(s) + \frac{2}{2s+1}u_2(s) + \frac{1}{2s+1}d(s)$$

$$W^u = \text{diag}(0.1, 0.1), W^y = 1, P = 10, M = 5$$

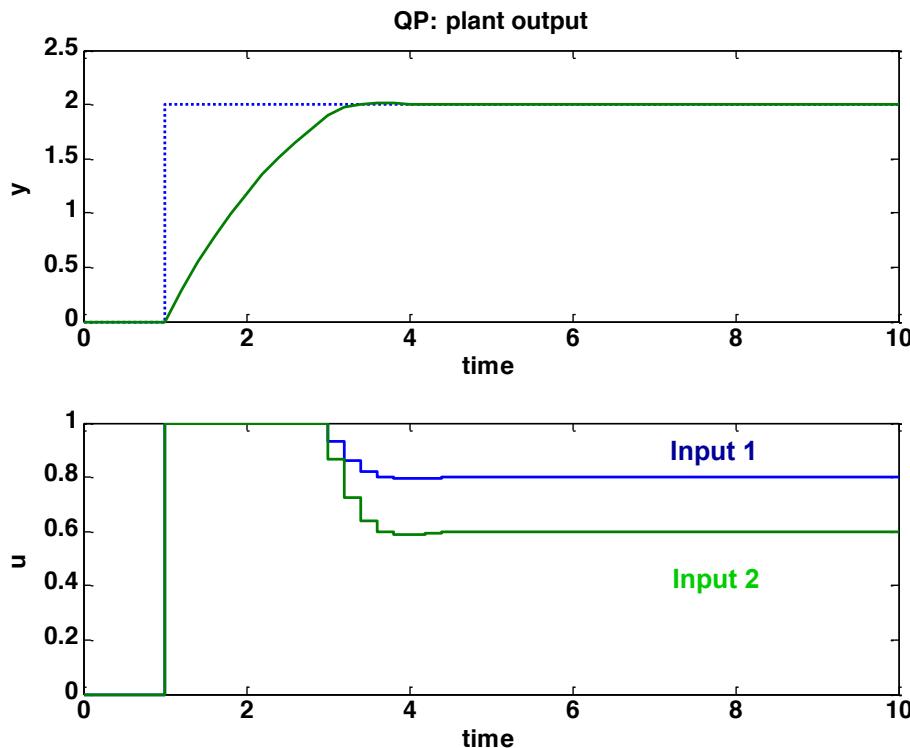


Inputs converge to different values

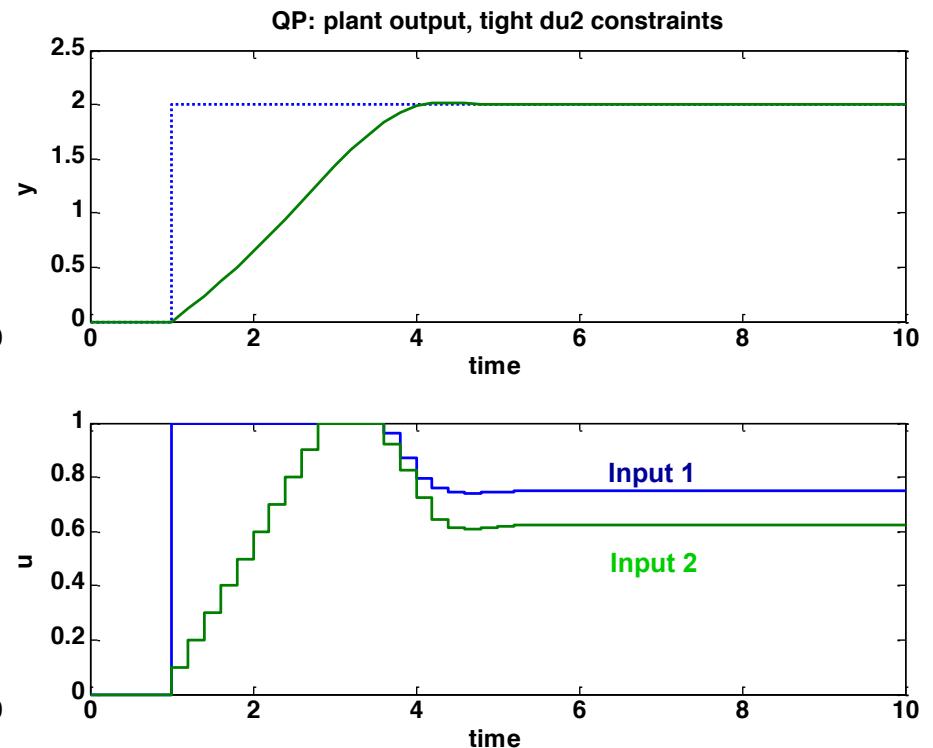
B. Wayne Bequette

2 input – 1 output

Setpoint Change, with constraints



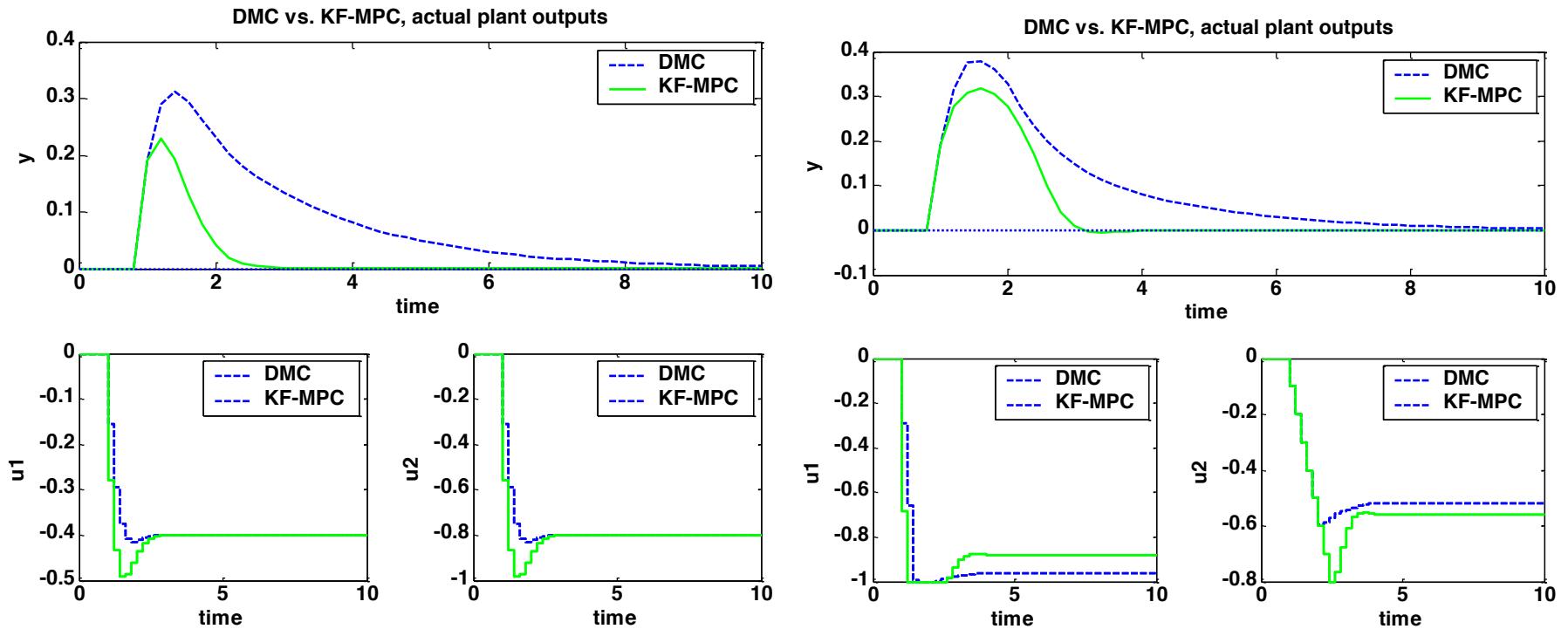
Constrained



Constrained, $\Delta u_2 \leq 0.1$

2 input – 1 output

Disturbance Rejection: unknown step in input

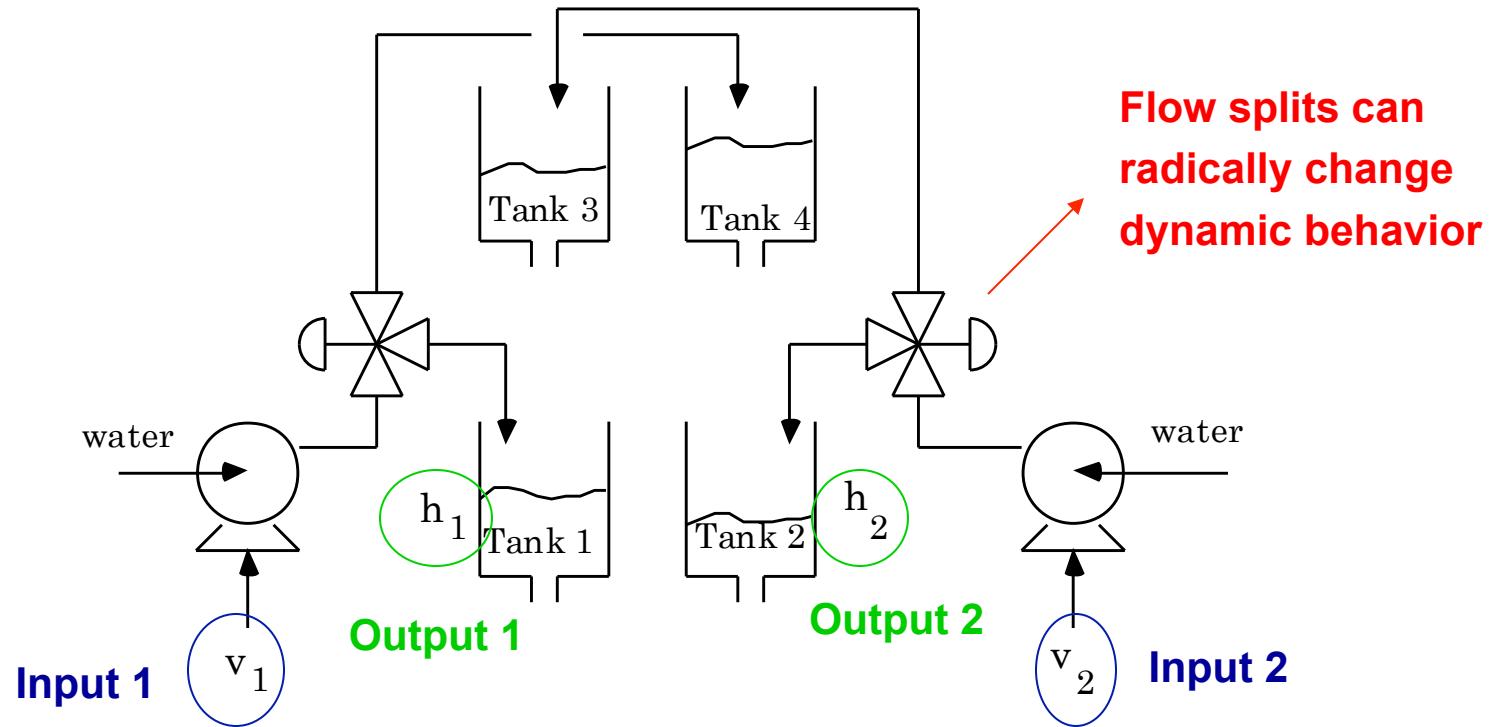


Tight velocity bounds (0.1) on u_1

2 inputs – 1 output

- Many input combinations can achieve same output in steady-state
 - Tuning/constraints determine input values
- “Habituating Control” idea can be used
 - Fast, but “expensive” input for immediate disturbance rejection
 - Slow, but “cheap” input for long-term compensation

Quadruple Tank Problem



Quadruple Tank Problem

Operating Point 1

$$G_1(s) = \begin{bmatrix} \frac{2.6}{62s+1} & \frac{1.5}{(23s+1)(62s+1)} \\ \frac{1.4}{(30s+1)(90s+1)} & \frac{2.8}{(90s+1)} \end{bmatrix}$$

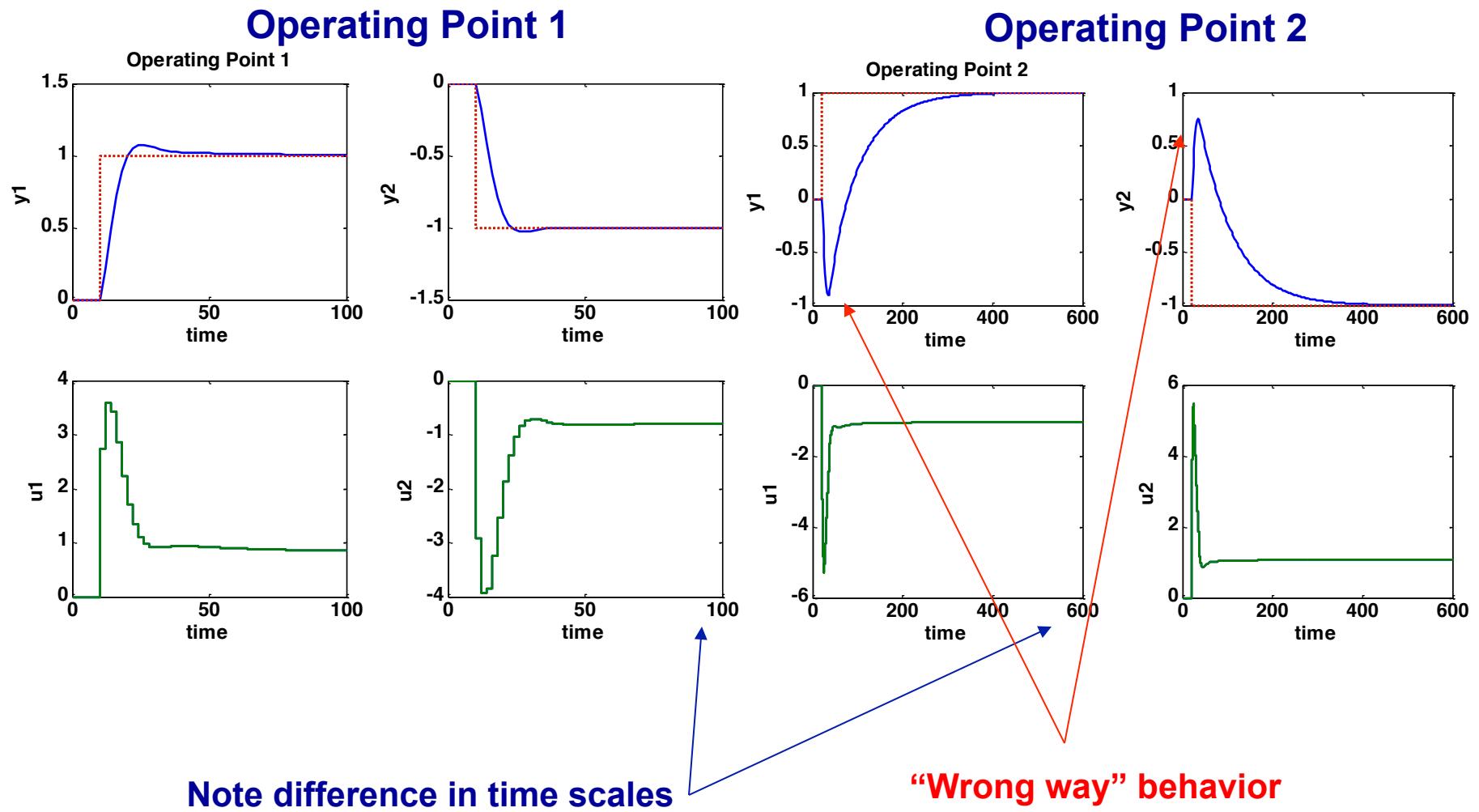
Operating Point 2

$$G_2(s) = \begin{bmatrix} \frac{1.5}{63s+1} & \frac{2.5}{(39s+1)(63s+1)} \\ \frac{2.5}{(56s+1)(91s+1)} & \frac{1.6}{(91s+1)} \end{bmatrix}$$

No time-delays or right-half-plane zeros.

Should we expect similar closed-loop performance?

Setpoint Responses



Quadruple Tank Problem

Operating Point 1 – Minimum Phase

$$G_1(s) = \begin{bmatrix} \frac{2.6}{62s+1} & \frac{1.5}{(23s+1)(62s+1)} \\ \frac{1.4}{(30s+1)(90s+1)} & \frac{2.8}{(90s+1)} \end{bmatrix} \quad z = -0.060 \text{ and } -0.018 \text{ sec}^{-1}$$

Operating Point 2 – Nonminimum Phase (RHPT zero)

$$G_2(s) = \begin{bmatrix} \frac{1.5}{63s+1} & \frac{2.5}{(39s+1)(63s+1)} \\ \frac{2.5}{(56s+1)(91s+1)} & \frac{1.6}{(91s+1)} \end{bmatrix} \quad z = -0.057 \text{ and } +0.013 \text{ sec}^{-1}$$


Multivariable Systems

- Can have right-half-plane “**transmission zeros**” even when no individual transfer function has a RHP zero
- Can have individual RHP zeros yet not have a RHPT zero
 - Fine performance when constraints are not active
 - May fail when one constraint becomes active or a loop is “opened”
- Can exhibit “**directional sensitivity**” – with some setpoint directions much easier to achieve than others
- Some of these MV properties cause challenges *independent of control strategy selected*

State Space Form not Limiting

- Step Response or Discrete TF can be written in State Space form
- Virtually all recent theoretical results are based on a state space formulation

Summary

- State Space MPC
- Disturbance Models
 - State/input or/and output disturbance
 - State/input form handles unstable systems
- Observer Design
 - Kalman Filter
- Can Control Outputs that are not measured
- Manipulated Input Blocking
 - Reduce number of decision variables

Stability Proofs: Linear Models

- Unconstrained
 - Infinite horizon, State and Input Weights
 - LQ Control
 - Couple with Kalman Filter if some states are unmeasured
 - State penalty at end of horizon
- Constrained
 - Terminal state constraint
 - Stabilizing state feedback at end of prediction horizon